

Today's Topics

Friday, February 6, 2026 (Week 2, lecture 7) – Chapter 3.

0. Gravity review

1. Circular Motion

... Newton's version of Kepler's 3rd law.

2. Center of Mass

3. Angular momentum

Problem Set #2 was due today at 9 am on ExpertTA... extended to 11:59 pm tonight.

Problem Set #3 – part 1 is due Friday, February 13 at 9 am on ExpertTA.

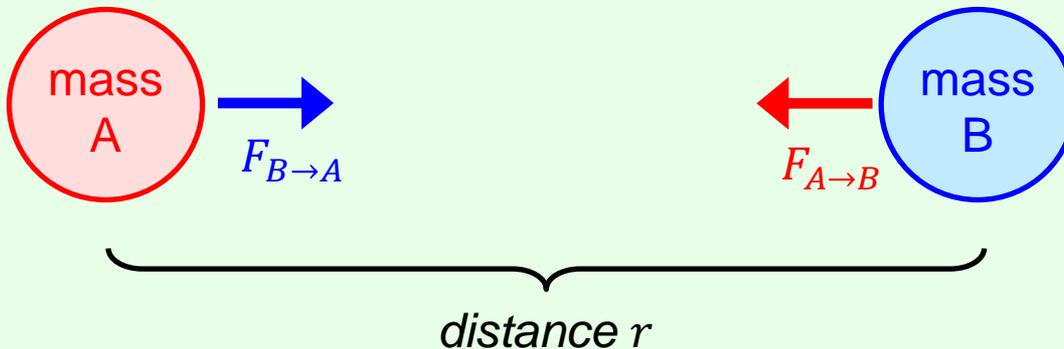
Problem Set #3 – part 2 is due Friday, February 13 at 9 am in-class (on paper).

Gravity Review

Newton's law of universal gravitation

All masses attract each other according to the following relation:

$$F_{A \rightarrow B} = -G \frac{M_A M_B}{r^2} = -F_{B \rightarrow A}$$



Properties

- Falls off as $1/r^2$.
- Proportional to M_A .
- Proportional to M_B .
- $G =$ Newton's constant
 $= 6.67430(15) \times 10^{-11}$
 $m^3 / Kg \cdot s^2$

Why do all objects (on Earth)
fall
at the same rate?

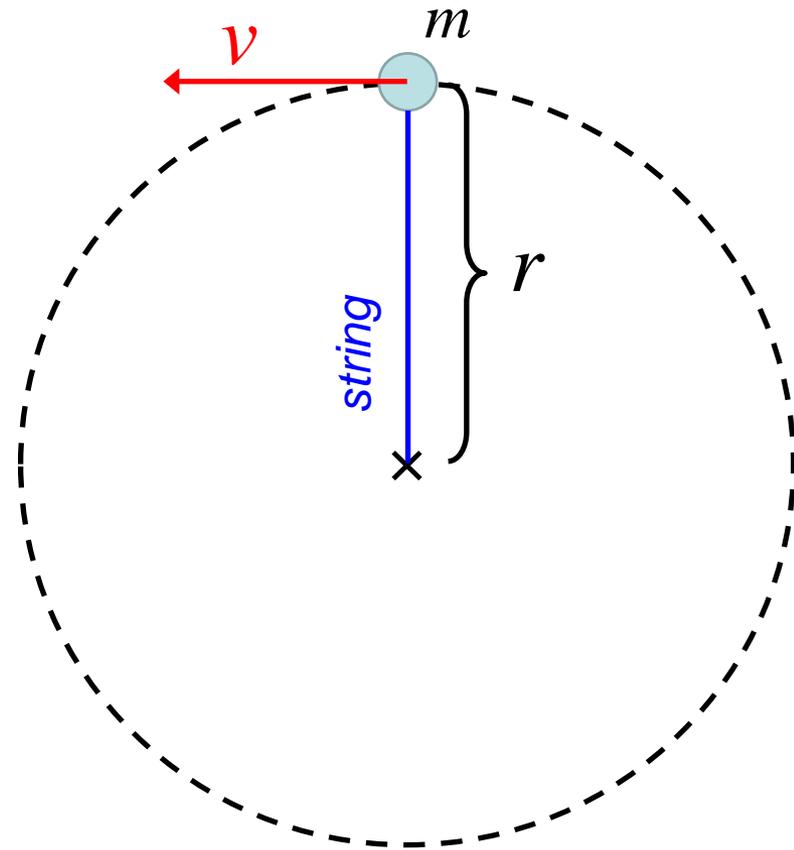
Circular Motion

Recall

acceleration = **change** in **velocity** over time

speed & direction

Rotation is a type of acceleration where the velocity **direction changes**, but speed is constant.



“ball on a string”

Circular Motion

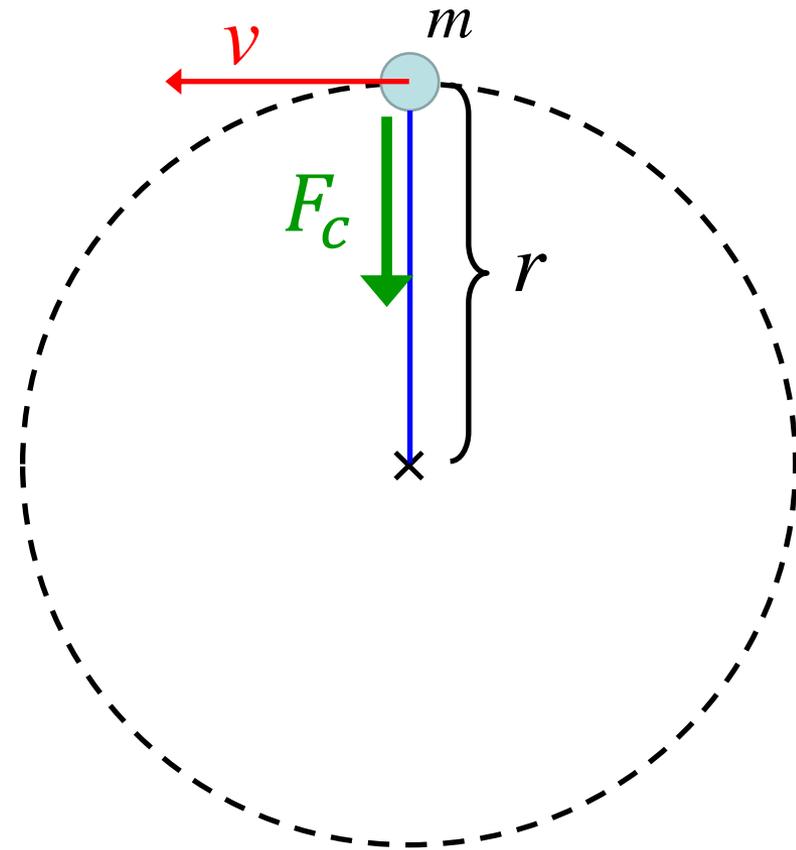
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Rotation is a type of acceleration where the velocity **direction changes**, but speed is constant.

$$\text{Acceleration: } a_c = \frac{v^2}{r}$$

$$\text{Centripetal Force: } F_c = \frac{mv^2}{r}$$

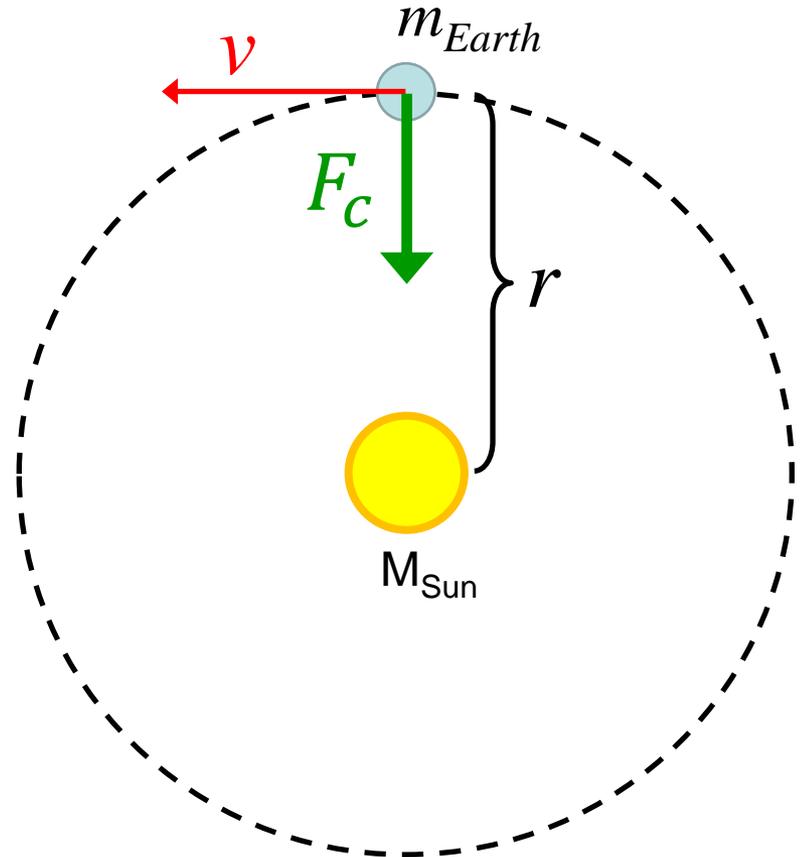


“ball on a string”

Circular Motion Example: Earth's orbit of Sun

Centripetal force needed to keep
Earth on a circular orbit:

$$F_c = \frac{m_{Earth} v_{Earth}^2}{r}$$



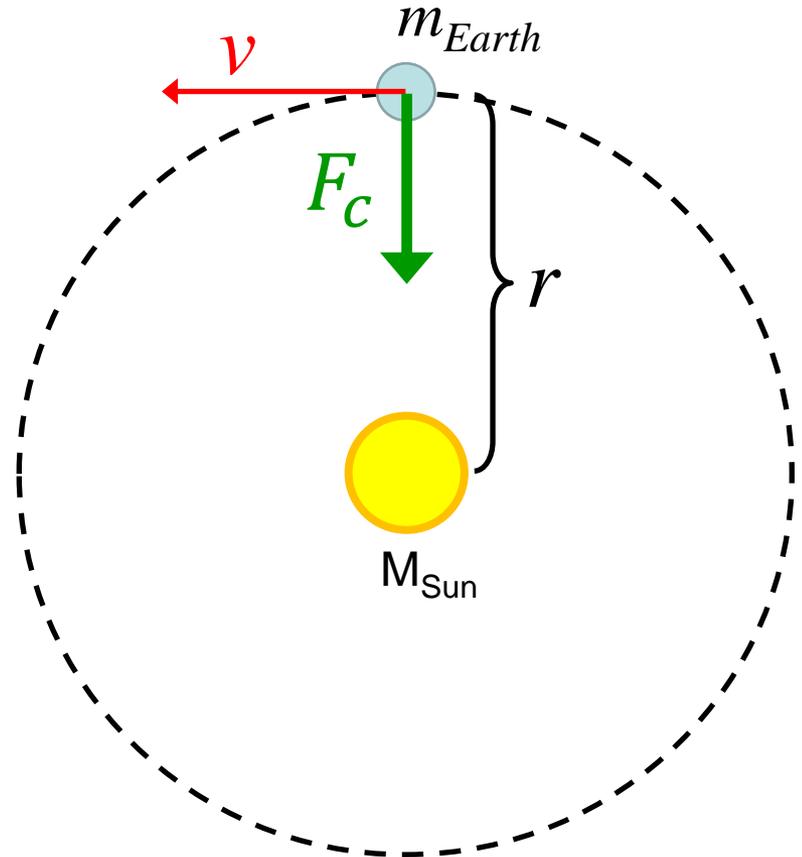
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$$F_{gravity, S \rightarrow E} = G \frac{m_{Earth} M_{Sun}}{r^2}$$



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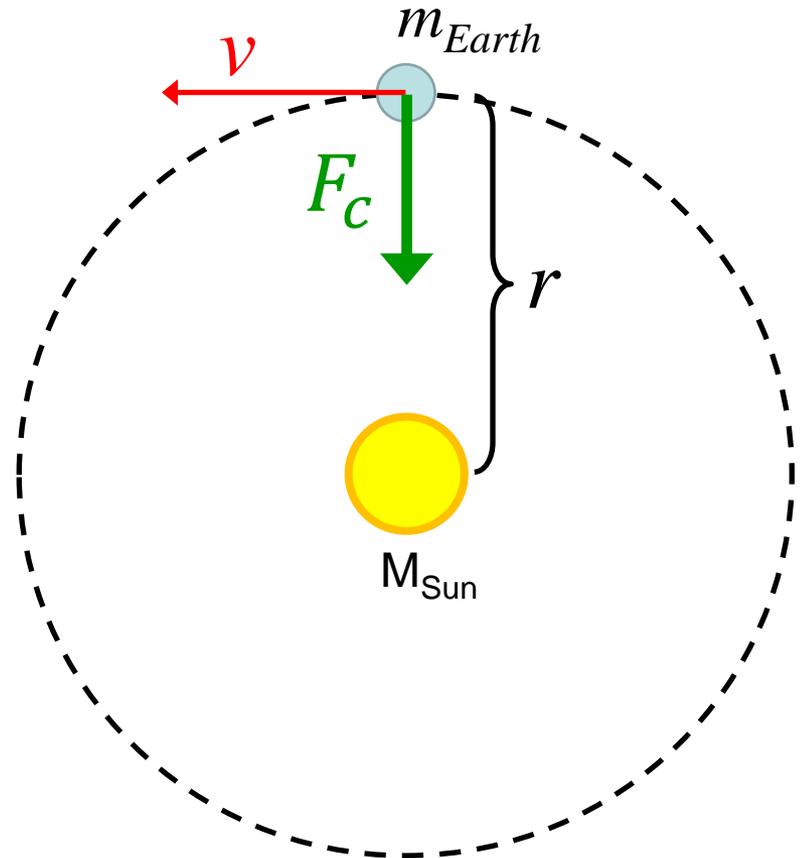
Force of gravity on Earth from Sun:

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The **centripetal force** that pulls on Earth to make it orbit the Sun **is gravity**:

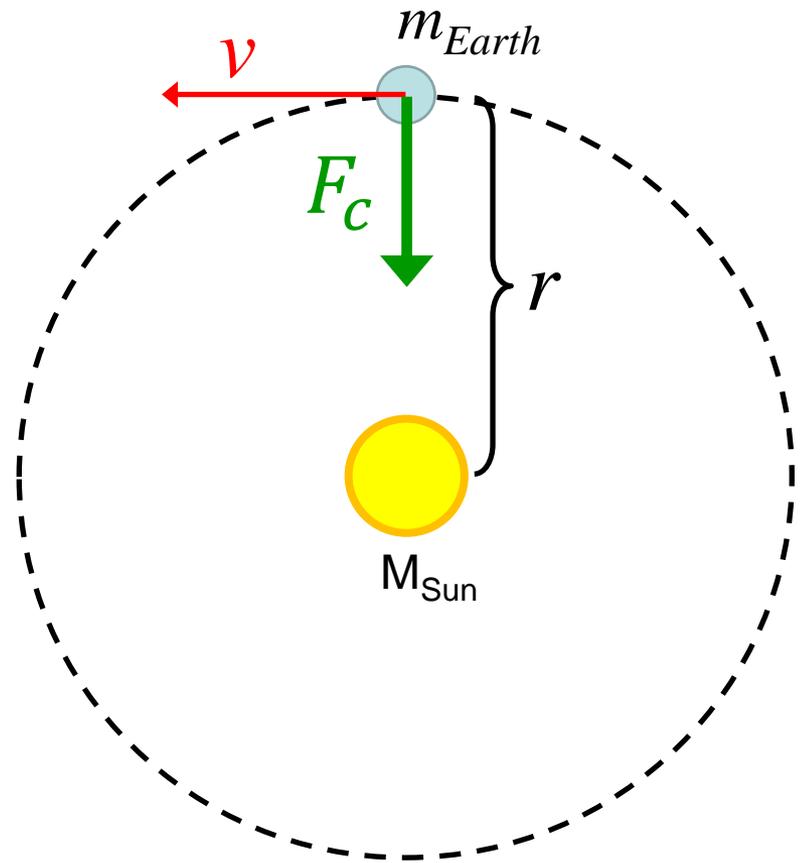
$$F_c = F_{gravity, S \rightarrow E}$$

$$\Leftrightarrow \frac{m_{Earth} v_{Earth}^2}{r} = G \frac{m_{Earth} M_{Sun}}{r^2}$$



Circular Motion Example: Earth's orbit of Sun

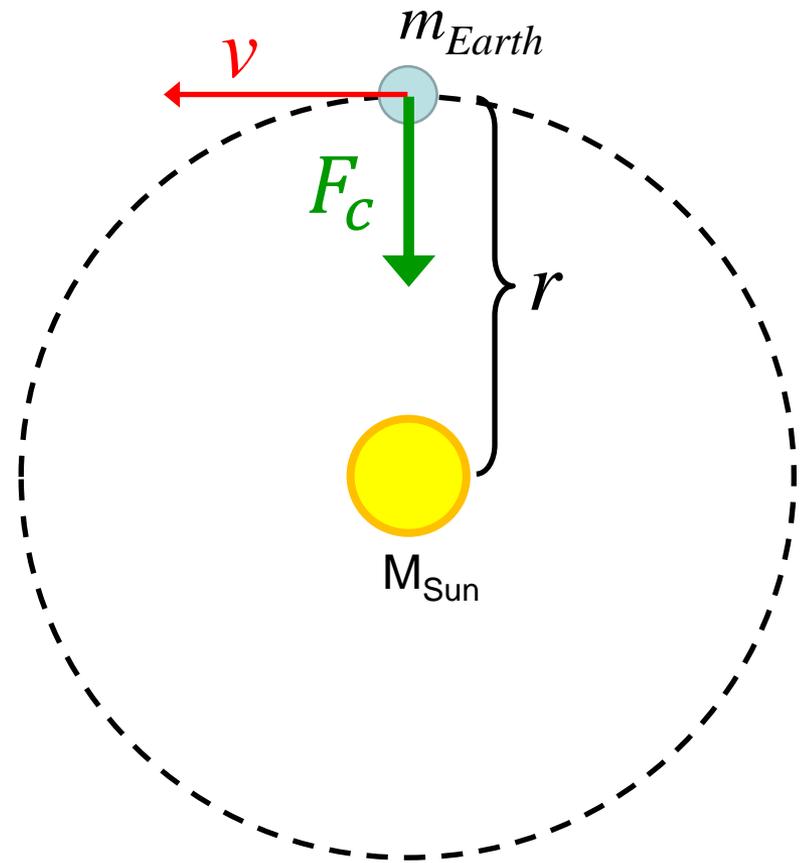
$$\frac{m_{\text{Earth}} v_{\text{Earth}}^2}{r} = G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r^2}$$



Circular Motion Example: Earth's orbit of Sun

$$\frac{\cancel{m_{\text{Earth}}} v_{\text{Earth}}^2}{r} = G \frac{\cancel{m_{\text{Earth}}} M_{\text{Sun}}}{r^2}$$

$$\Leftrightarrow v_{\text{Earth}}^2 = G \frac{M_{\text{Sun}}}{r}$$



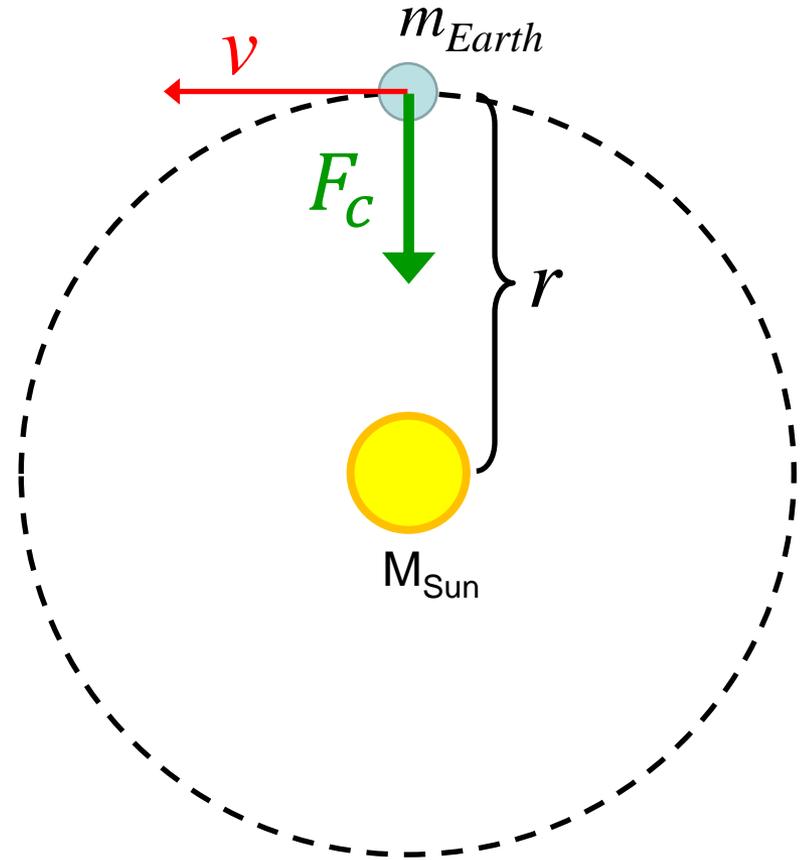
Circular Motion Example: Earth's orbit of Sun

$$\frac{\cancel{m_{Earth}} v_{Earth}^2}{r} = G \frac{\cancel{m_{Earth}} M_{Sun}}{r^2}$$

$$\Leftrightarrow v_{Earth}^2 = G \frac{M_{Sun}}{r}$$

Solve for M_{Sun} :

$$M_{Sun} = \frac{r v_{Earth}^2}{G}$$



Circular Motion Example: Earth's orbit of Sun

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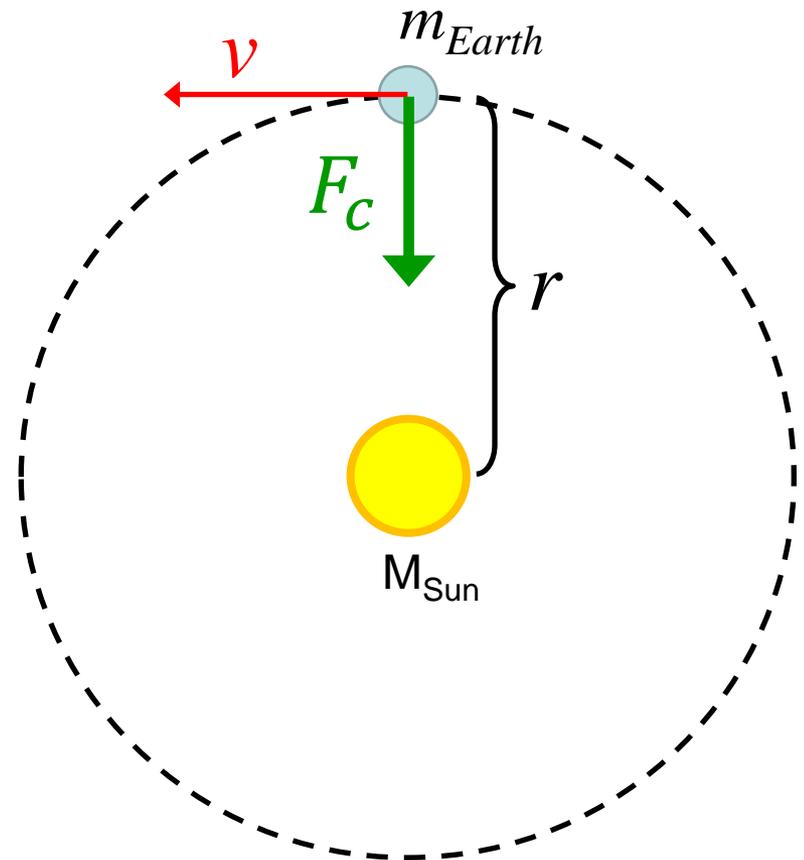
Solve for M_{Sun} :

$$M_{Sun} = \frac{r v_{Earth}^2}{G}$$

$$v_{Earth} = 29.78 \times 10^3 \text{ m/s}$$

$$r = 1 \text{ AU} = 149.6 \times 10^9 \text{ m}$$

$$G = 6.67430(15) \times 10^{-11} \text{ m}^3/\text{Kg} \cdot \text{s}^2$$



Circular Motion Example: Earth's orbit of Sun

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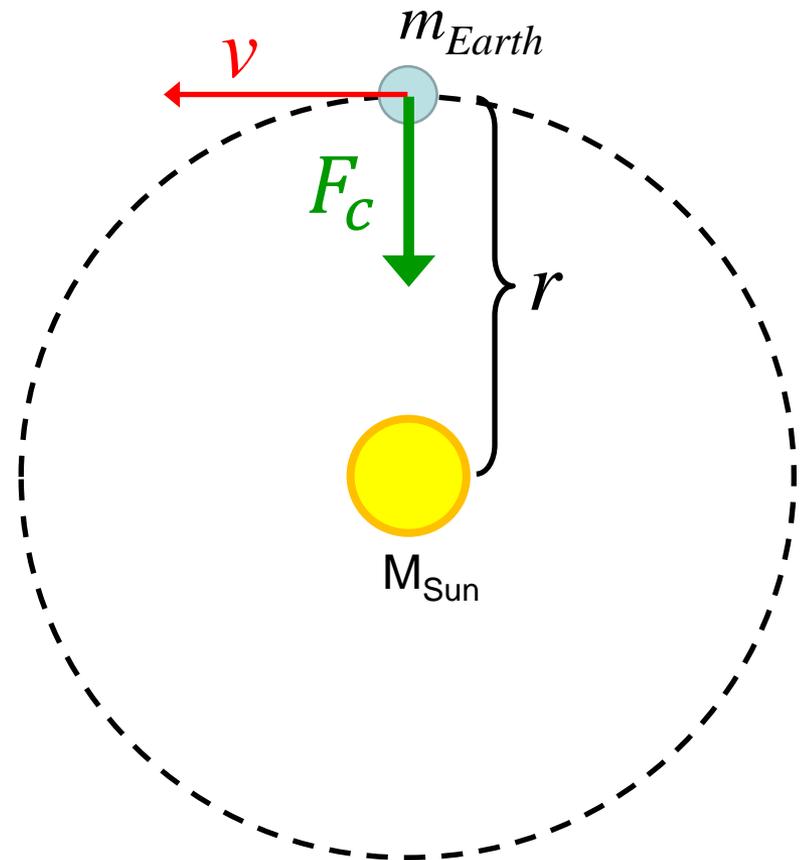
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$$M_{Sun} = 1.988 \times 10^{30} \text{ Kg}$$



You can get the mass of the Sun from
Earth's orbital parameters !!!

Newton's version of Kepler's 3rd Law

$$T^2 = \frac{4\pi^2}{G(M_1 + M_2)} a^3$$

This formula is in SI units

T = orbital period in seconds

$M_{1,2}$ = Mass of orbiting objects in Kg

a = semimajor axis in meters

G = $6.6743 \times 10^{-11} \text{ m}^3/\text{Kg}\cdot\text{s}^2$

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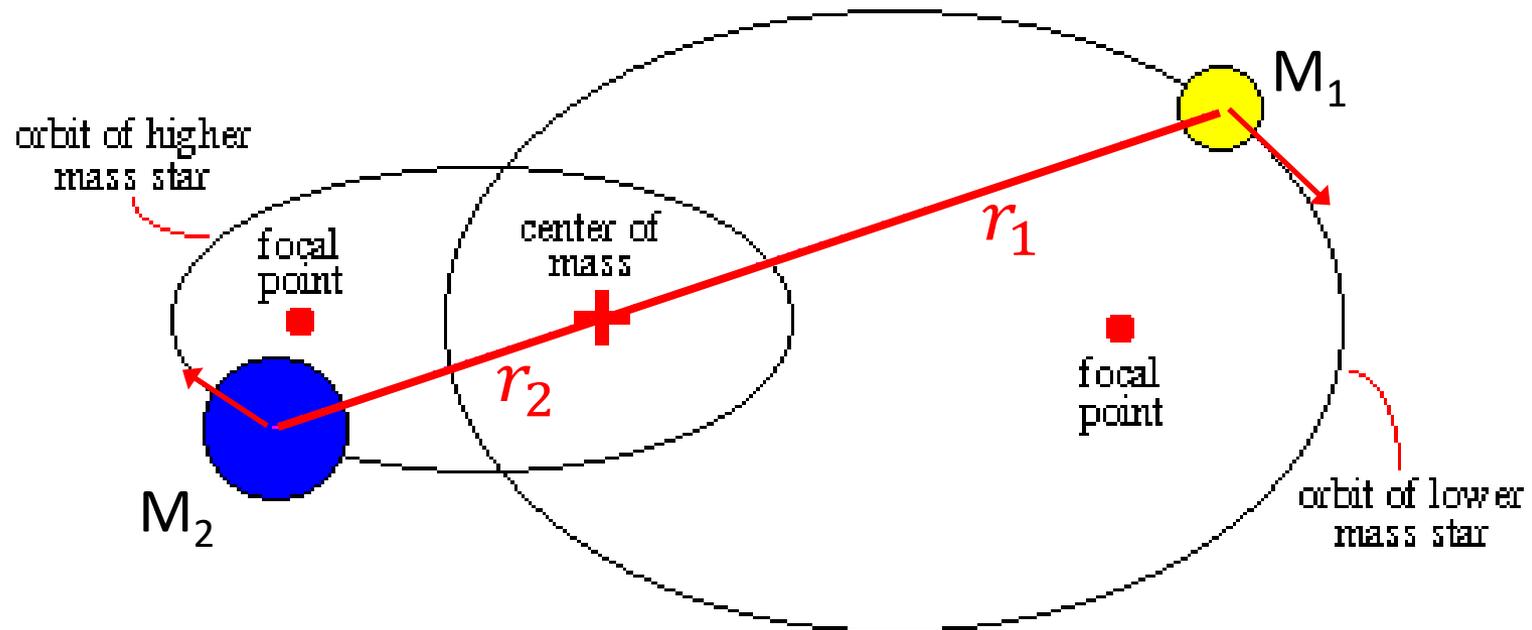
G = $6.6743 \times 10^{-11} \text{ m}^3/\text{Kg}\cdot\text{s}^2$

WHAT IF: What happens to the orbits if M_1 and M_2 are comparable ?

Center of Mass

What happens when $M_1 \simeq M_2$?

The **center of mass** of M_1 and M_2 serves as the orbiting ellipse focus.



[adapted from <http://abyss.uoregon.edu>]

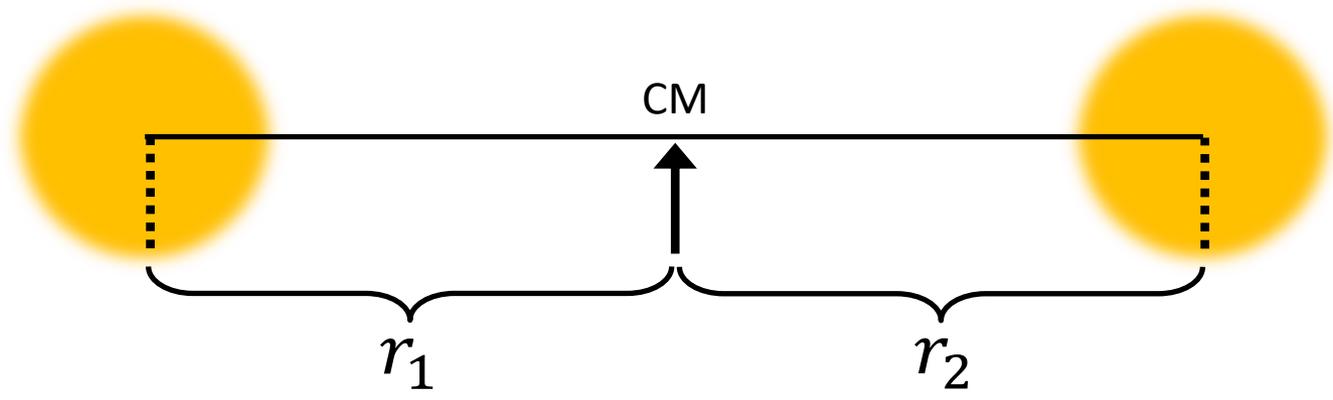
Semimajor axis “a”:

The coordinate “ $r = r_1 + r_2$ ” is the distance between the two masses. It also describes an ellipse (not shown), whose semimajor axis “a” is used in Newton’s version of Kepler’s 3rd law.

Center of Mass

mass M_1

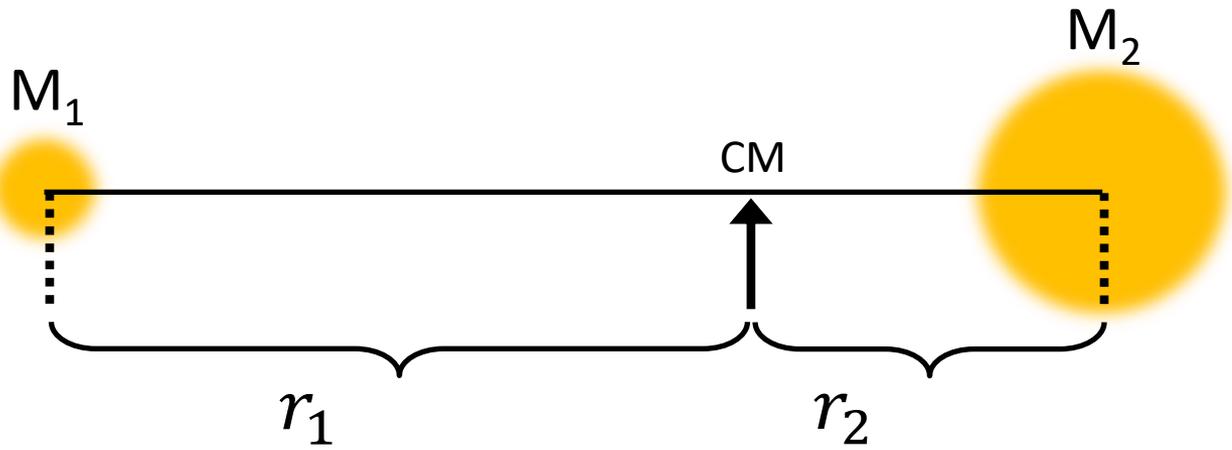
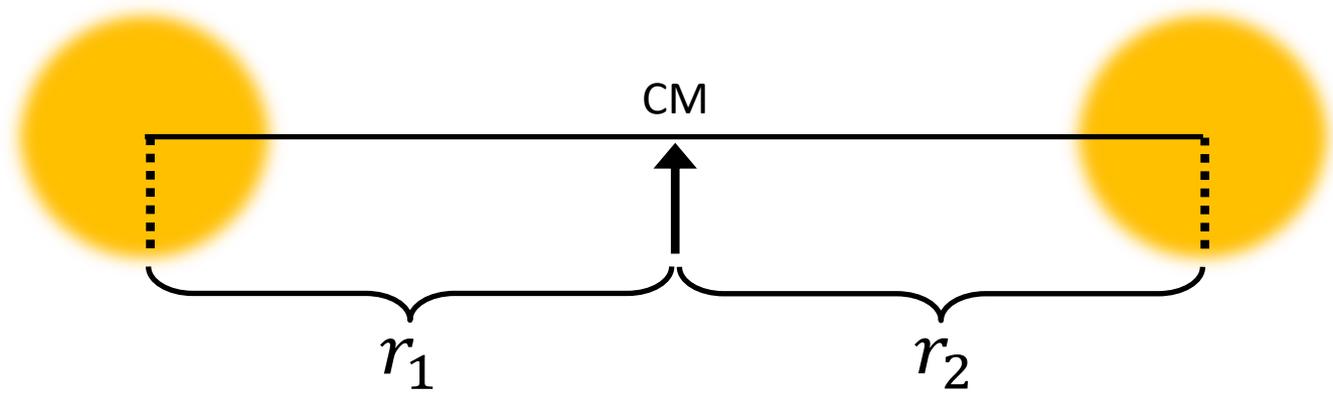
mass M_2



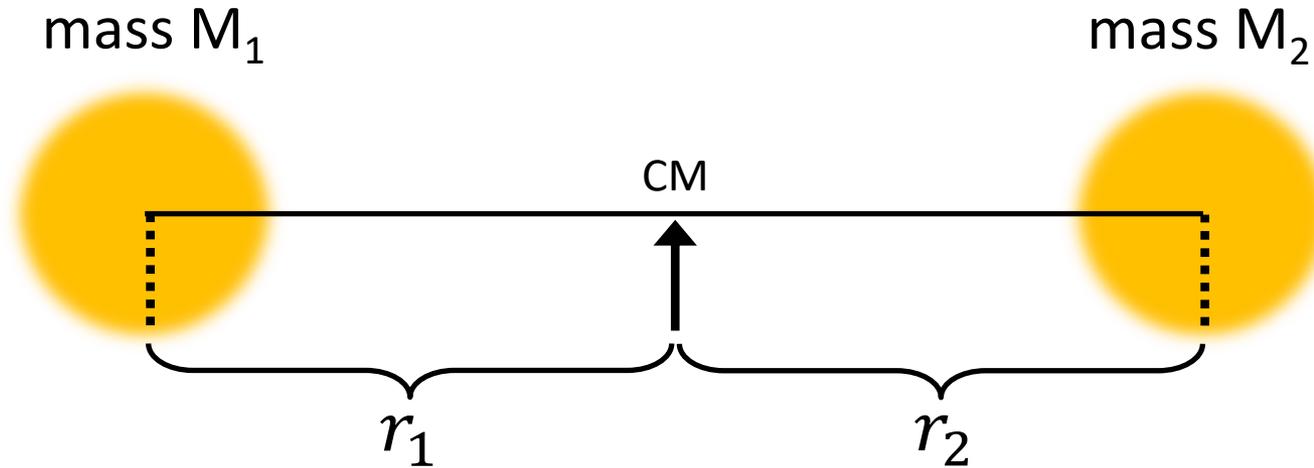
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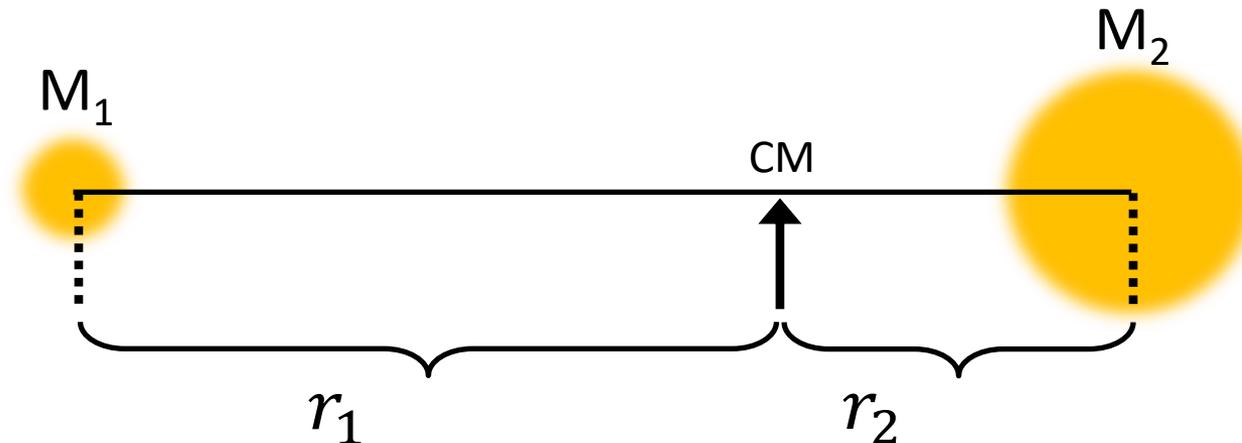
mass M_2



Center of Mass



Center of mass is located such that $M_1 r_1 = M_2 r_2$
(or "barycenter")



Some Barycenters

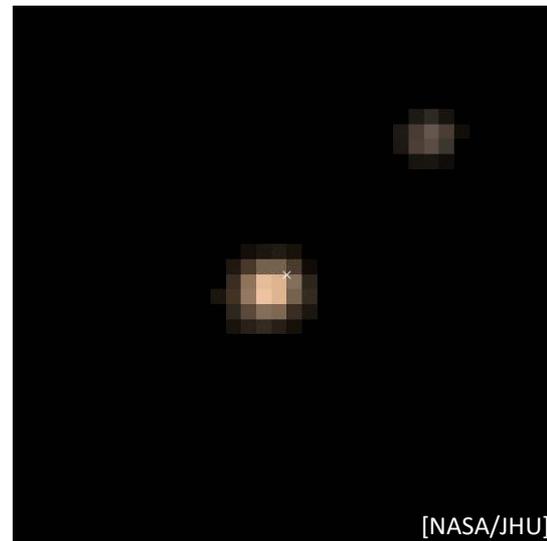
$$M_2 - M_1: \quad r_2 = a \frac{M_1}{M_1 + M_2} = \text{distance from CM to } M_2$$

$$\text{Sun-Earth:} \quad r_2 = 448 \text{ km} = 3.0 \times 10^{-6} \text{ AU}$$

$$\text{Earth-Moon:} \quad r_2 = 4,670 \text{ km with } a = 384,000 \text{ km} \\ = 73\% \text{ of Earth's radius}$$

Pluto – Charon:

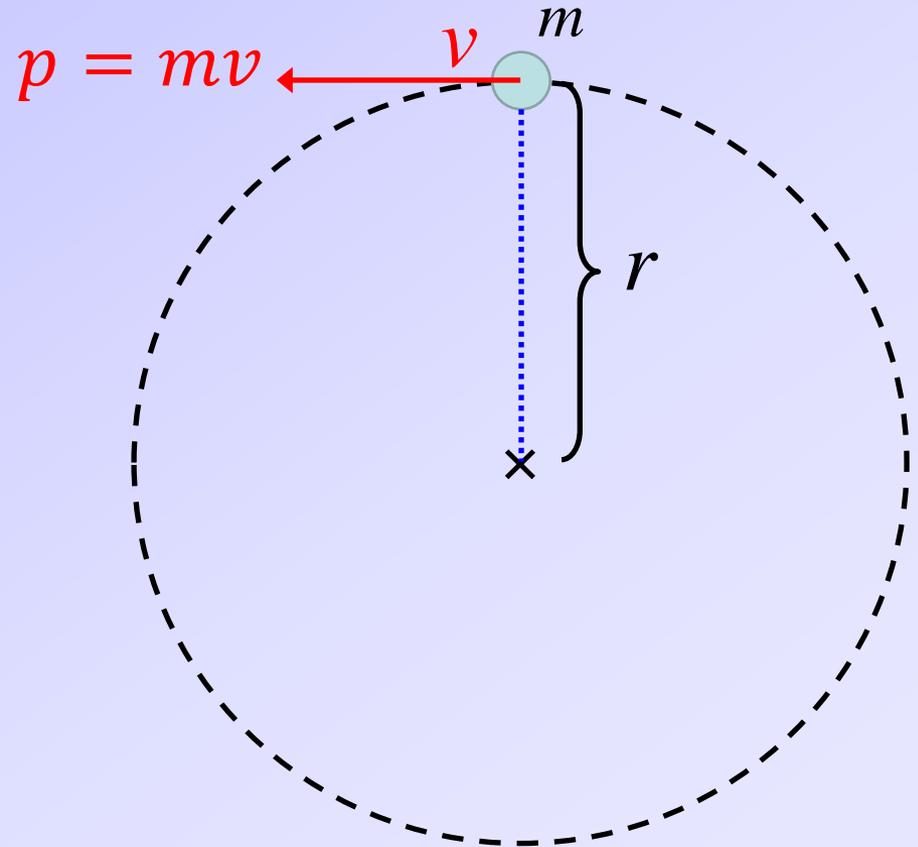
orbital period $T = 6.4$ days



PolleEv Quiz: PolleEv.com/sethaubin

Conservation of Angular Momentum (1)

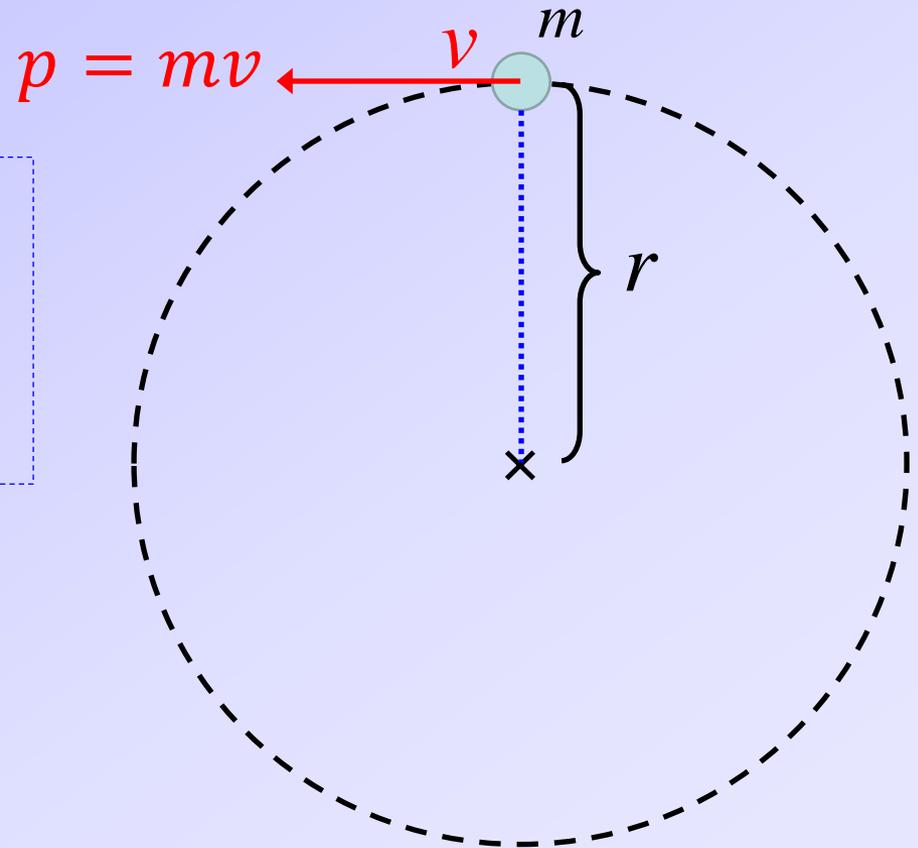
angular momentum = $L = \text{momentum} \times \text{radius}$
 $= p \times r \quad \dots = mvr$ for circular motion



Conservation of Angular Momentum (1)

angular momentum = L = momentum \times radius
= $p \times r$... = mvr for circular motion

total angular momentum
=
sum of the angular momenta of
all the sub-parts of a system



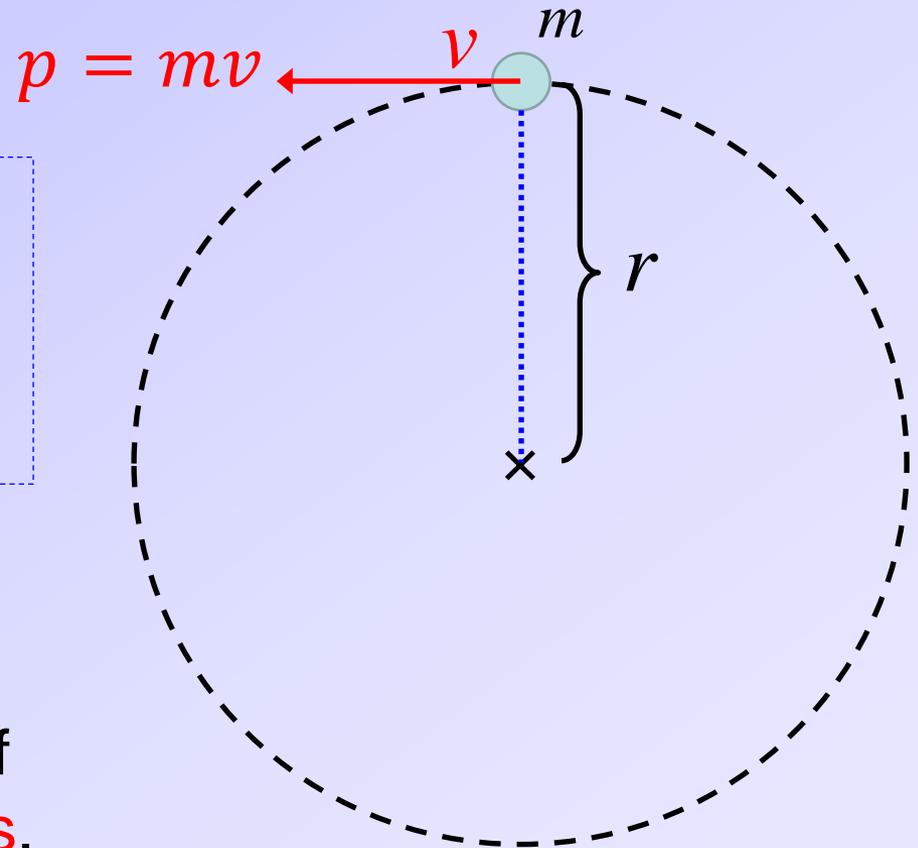
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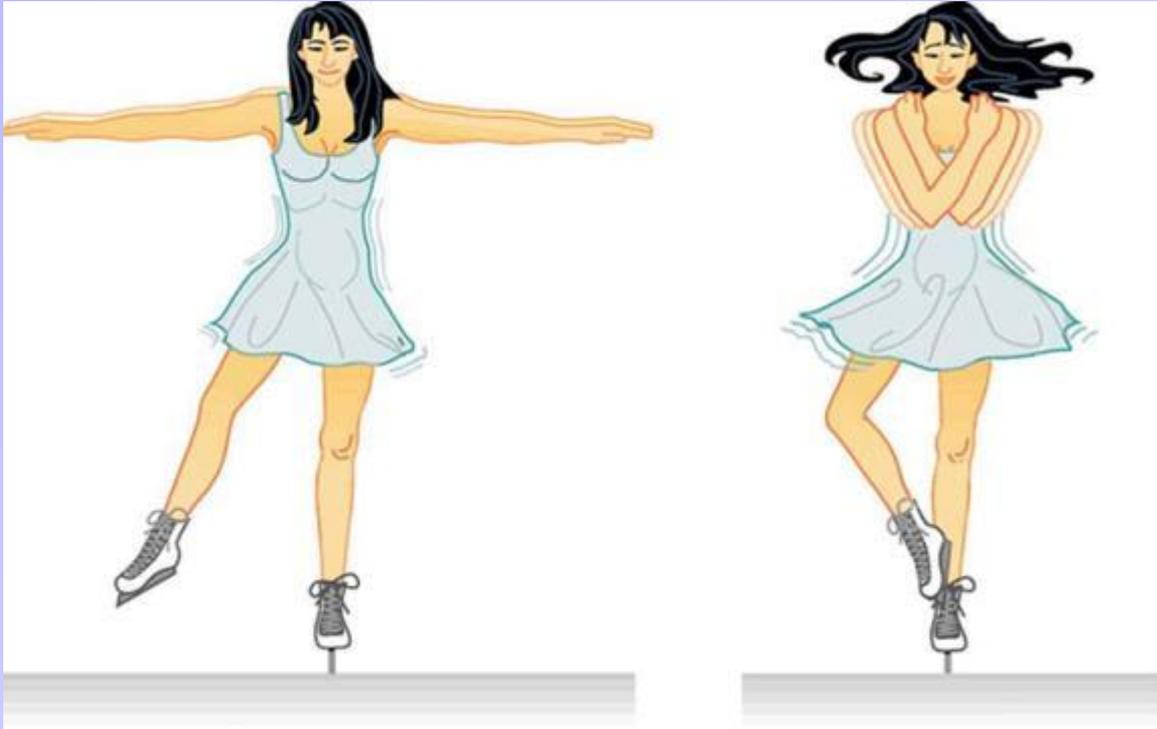
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sum of the angular momenta of
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Conservation Law

The **total angular momentum** of
a **closed system** **never changes**.



Conservation of Angular Momentum (2)



[OpenStax: Astronomy]

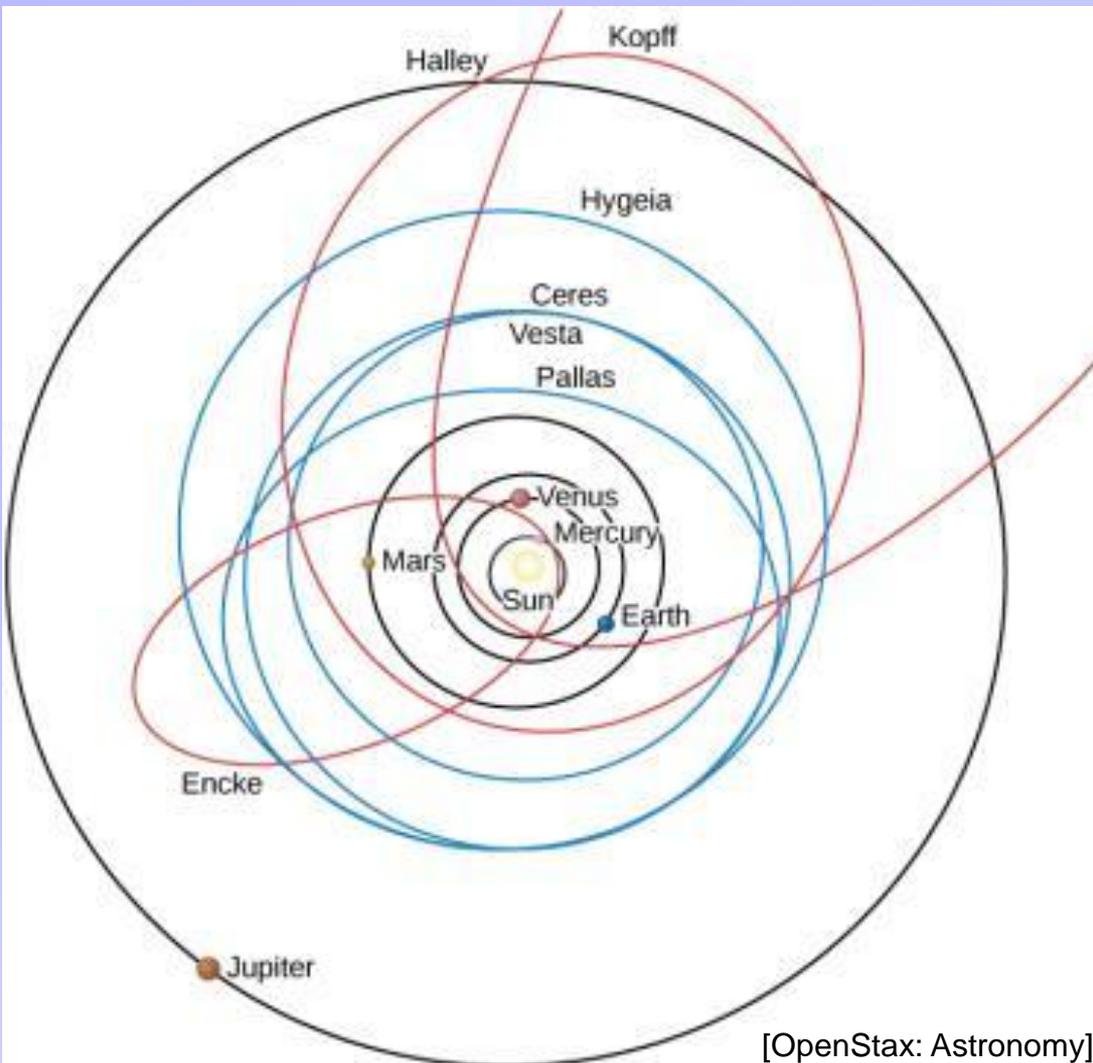
- When a spinning figure skater **brings in her arms**, their distance from her spin center is **smaller**, so her **speed increases**.
- When her **arms are out**, their distance from the spin center is **greater**, so she **slows down**.

Conservation of Angular Momentum

The multiple planets, asteroids, and comets all interact and modify each others orbits.

→ **Individual angular momenta change.**

→ **Total angular momentum of Solar System is constant.**



[OpenStax: Astronomy]

Planets (black), asteroids (blue), comets (red)

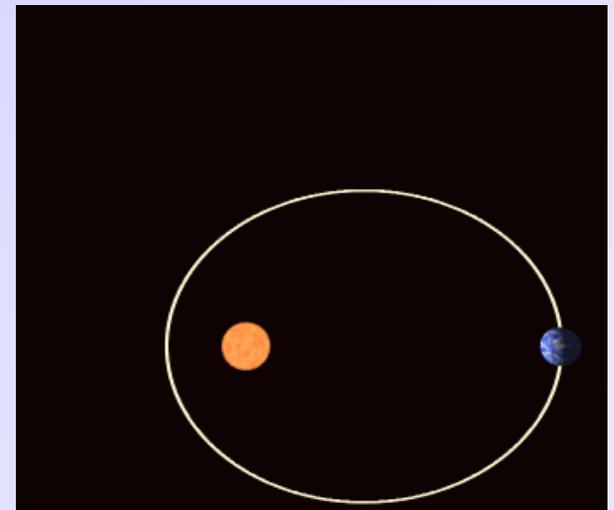
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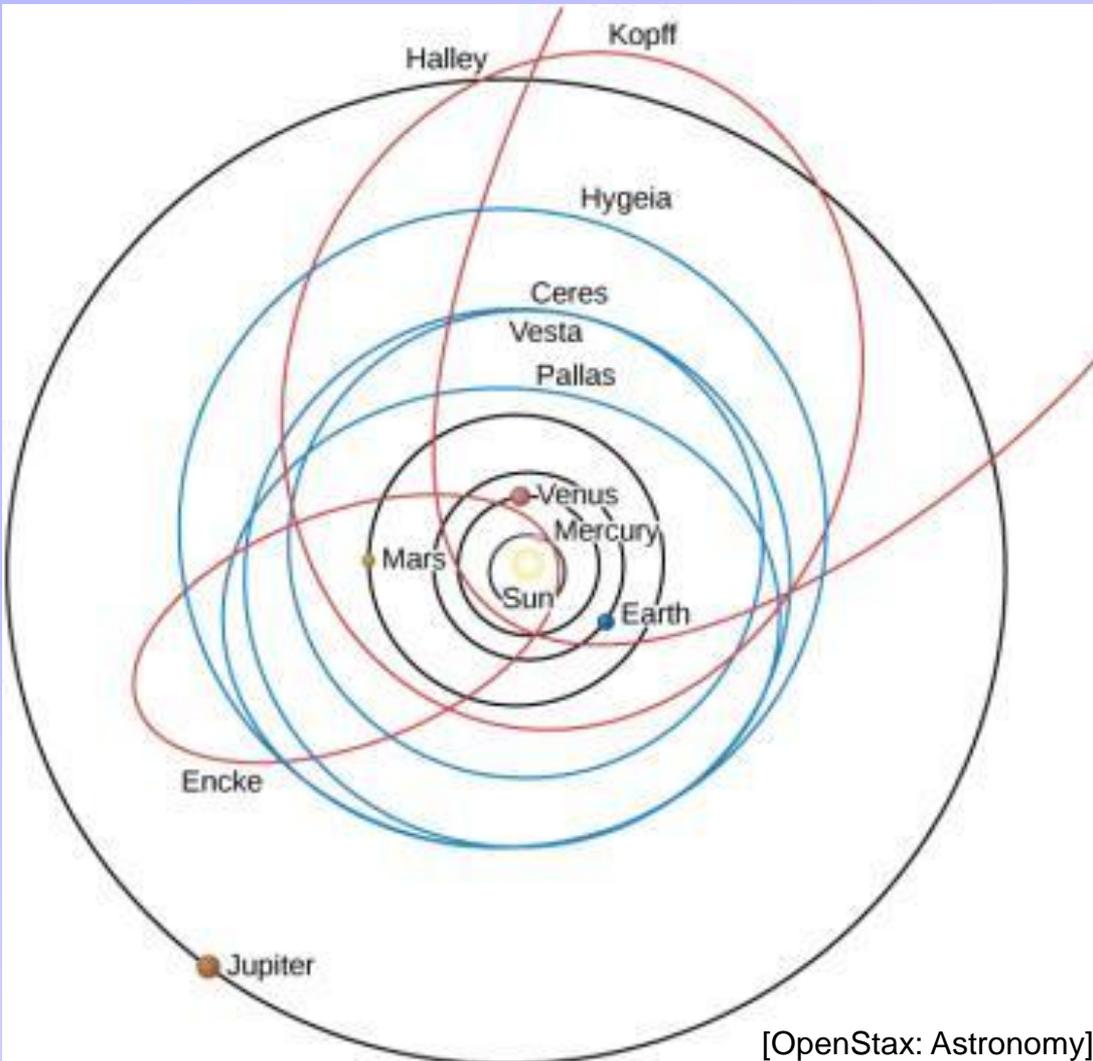
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Example: Apsidal Precession



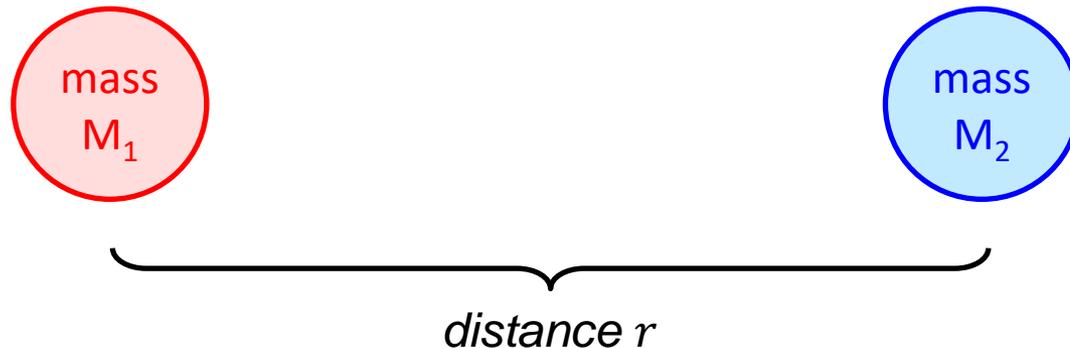
By WillowW - Own work, CC BY 3.0,
<https://commons.wikimedia.org/w/index.php?curid=3416065>



[OpenStax: Astronomy]

Planets (black), asteroids (blue), comets (red)

Gravitational Potential Energy



$$\text{Stored gravitational energy} = E_{\text{potential}} = -G \frac{M_1 M_2}{r}$$

$$\text{Total Energy} = E_{\text{total}} = E_{\text{potential}} + E_{\text{kinetic}}$$

For 2 orbiting bodies (e.g. Sun + Earth): $E_{\text{total}} < 0$

For 2 unbound bodies (Earth + Mars rocket): $E_{\text{total}} > 0$

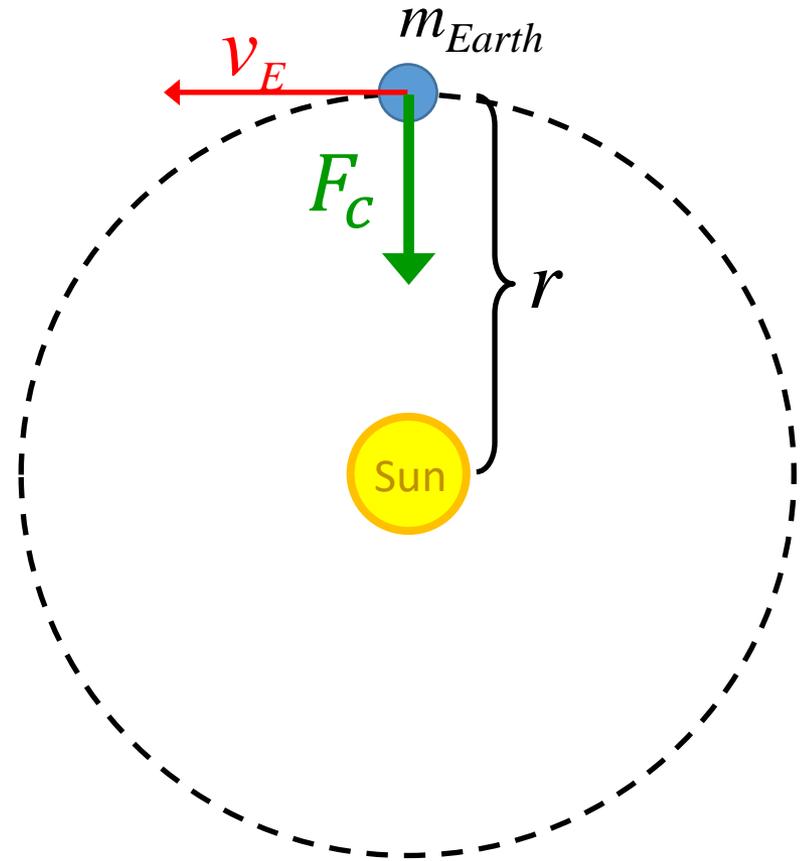
Bound Orbital Energy

Centripetal force needed to keep Earth on a circular orbit:

$$F_c = \frac{m_{\text{Earth}} v_{\text{Earth}}^2}{r}$$

Force of gravity on Earth from Sun:

$$F_{\text{gravity}, S \rightarrow E} = G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r^2}$$



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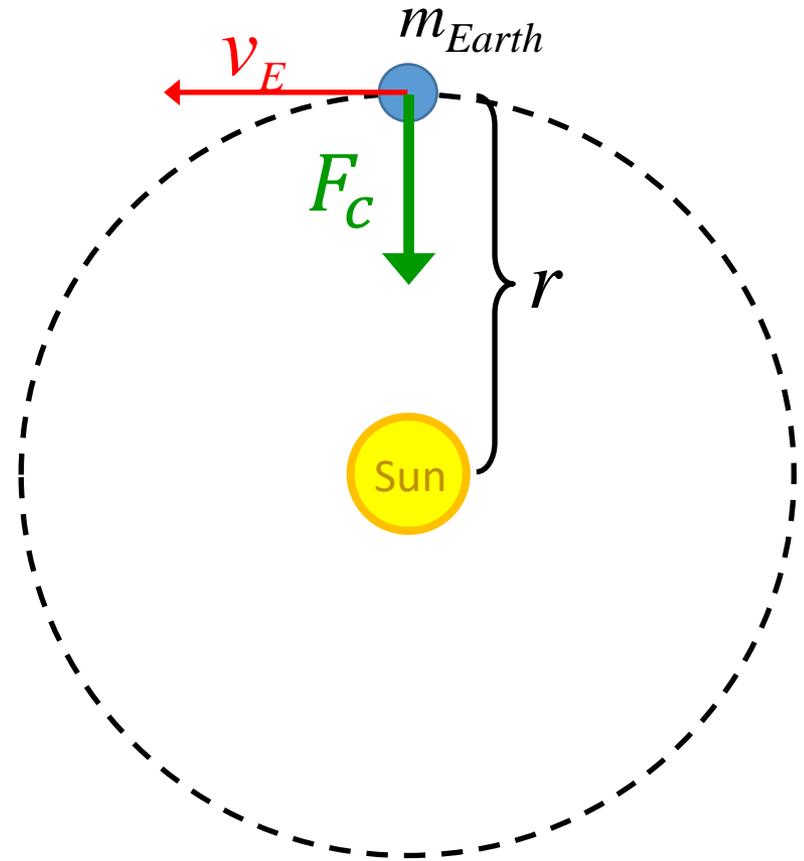
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The **centripetal force** that pulls on Earth to make it orbit the Sun **is gravity**:

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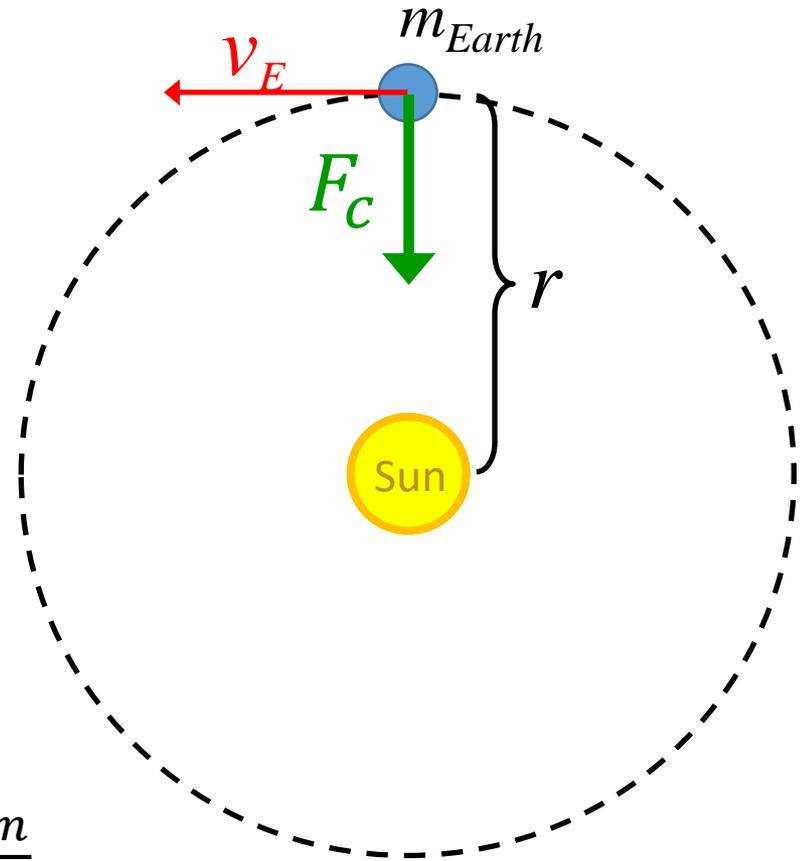
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$$\Leftrightarrow \frac{m_{\text{Earth}} v_{\text{Earth}}^2}{r} = G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r^2}$$

$$\Leftrightarrow \frac{1}{2} m_{\text{Earth}} v_{\text{Earth}}^2 = \frac{1}{2} G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r}$$



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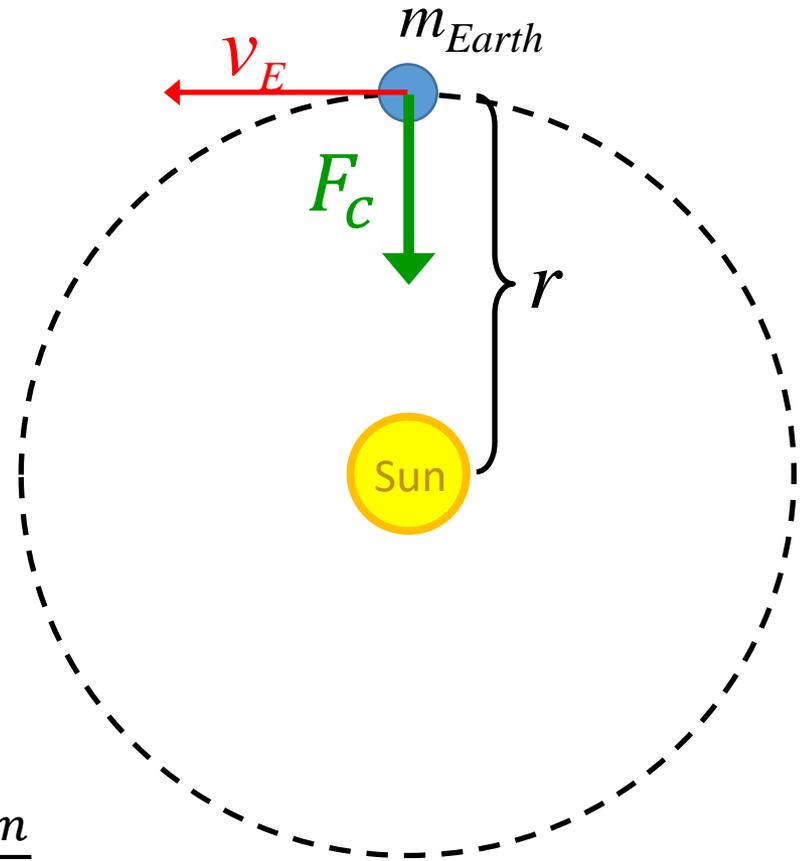
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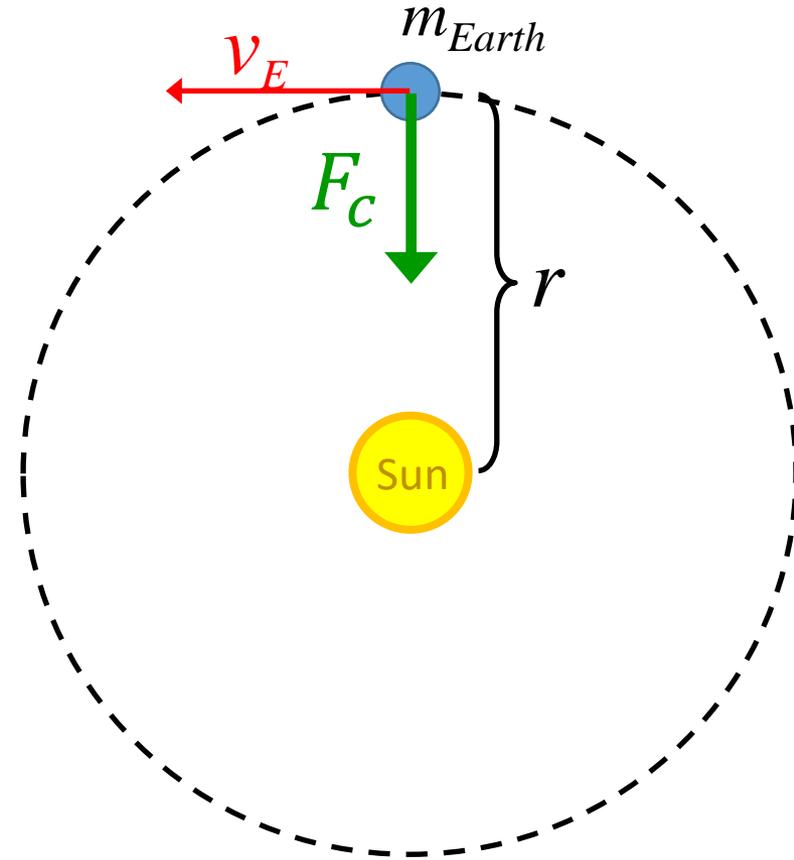
$$\Leftrightarrow \frac{1}{2} m_{\text{Earth}} v_{\text{Earth}}^2 = \frac{1}{2} G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r}$$

$$\Leftrightarrow E_{\text{kinetic}} = \frac{1}{2} m_{\text{Earth}} v_{\text{Earth}}^2 = \frac{1}{2} G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r}$$



Bound Orbital Energy

$$E_{kinetic} = \frac{1}{2} G \frac{m_{Earth} M_{Sun}}{r}$$

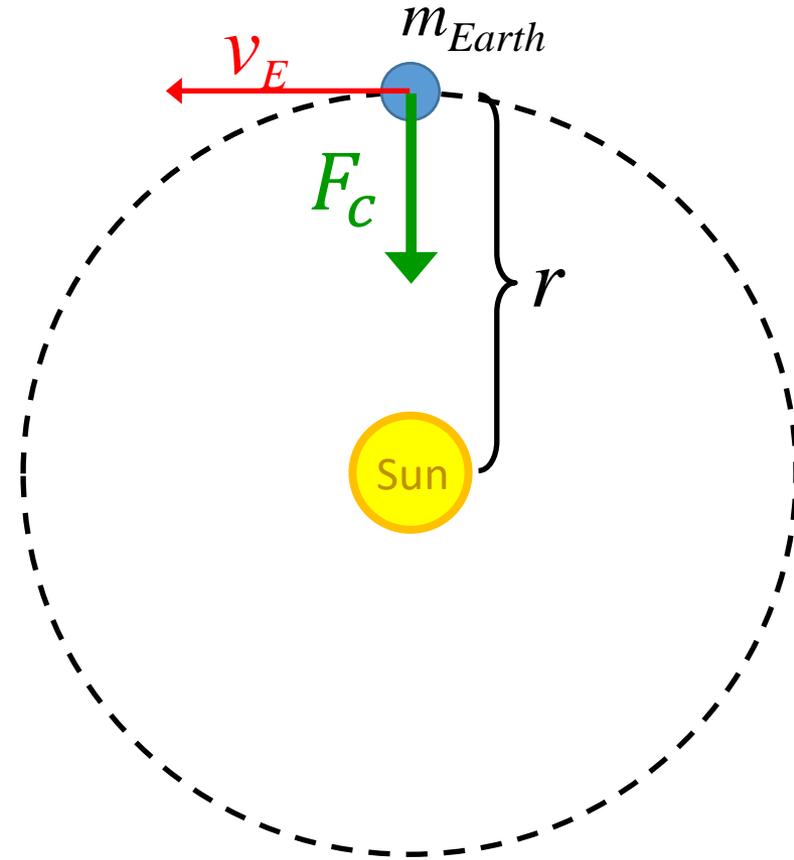


Bound Orbital Energy

$$E_{kinetic} = \frac{1}{2} G \frac{m_{Earth} M_{Sun}}{r}$$

Total Energy:

$$E_{Total} = E_{kinetic} + E_{potential}$$

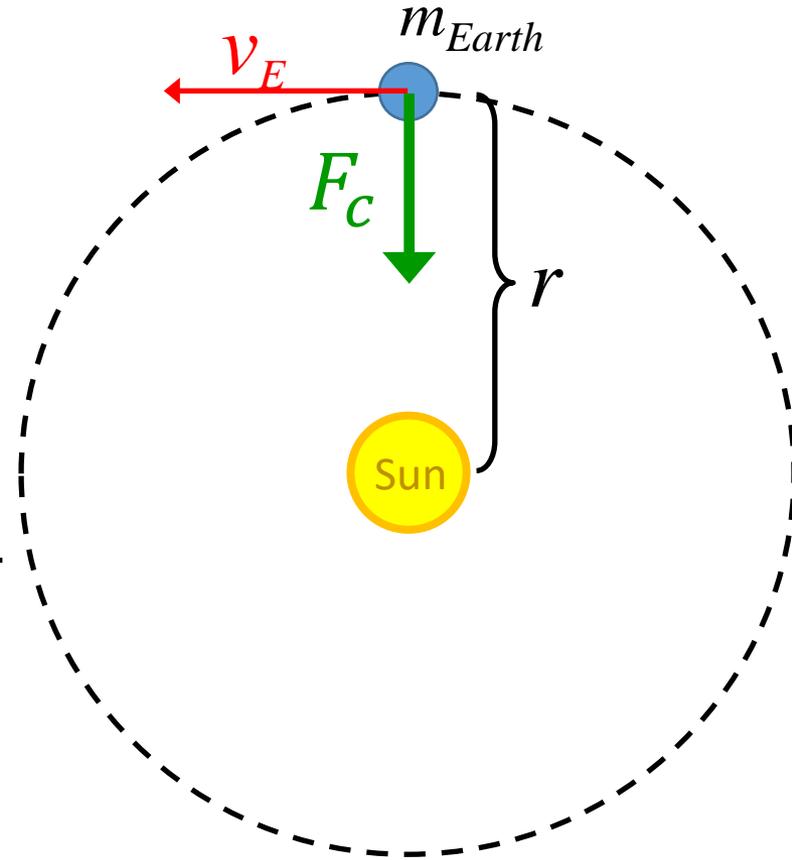


Bound Orbital Energy

$$E_{kinetic} = \frac{1}{2} G \frac{m_{Earth} M_{Sun}}{r}$$

Total Energy:

$$\begin{aligned} E_{Total} &= E_{kinetic} + E_{potential} \\ &= \frac{1}{2} G \frac{m_{Earth} M_{Sun}}{r} - G \frac{m_{Earth} M_{Sun}}{r} \end{aligned}$$



Bound Orbital Energy

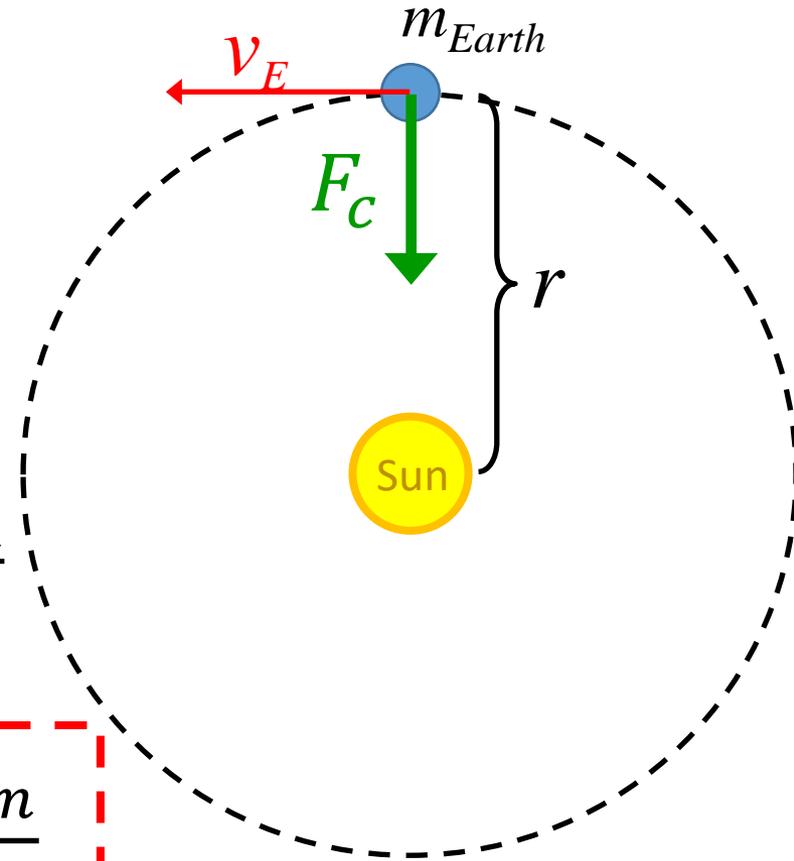
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Bound Orbital Energy

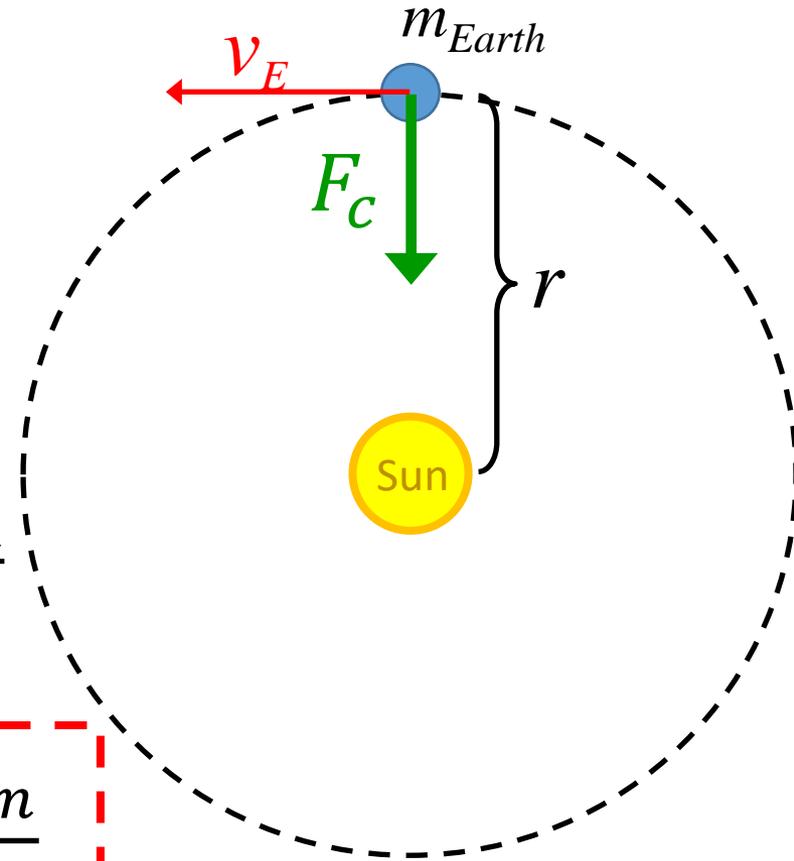
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$$\Leftrightarrow E_{Total} = -\frac{1}{2} G \frac{m_{Earth} M_{Sun}}{r}$$



The bound orbital energy is negative: $E_{Total} < 0$

Example: When a rocket wants to orbit another planet it has to slow (lower its energy) in order to go into orbit.