

Today's Topics

Friday, February 6, 2026 (Week 2, lecture 7) – Chapter 3.

0. Gravity review

1. Circular Motion

... Newton's version of Kepler's 3rd law.

2. Center of Mass

3. Angular momentum

Problem Set #2 was due today at 9 am on ExpertTA... extended to 11:59 pm tonight.

Problem Set #3 – part 1 is due Friday, February 13 at 9 am on ExpertTA.

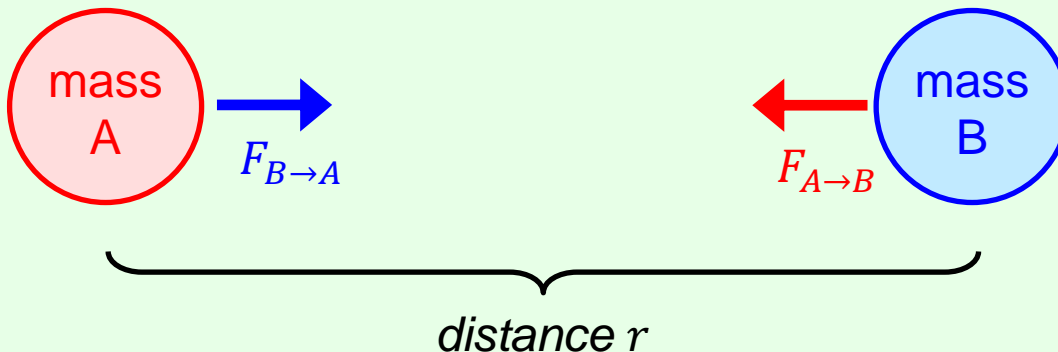
Problem Set #3 – part 2 is due Friday, February 13 at 9 am in-class (on paper).

Gravity Review

Newton's law of universal gravitation

All masses attract each other according to the following relation:

$$F_{A \rightarrow B} = -G \frac{M_A M_B}{r^2} = -F_{B \rightarrow A}$$



Properties

- Falls off as $1/r^2$.
- Proportional to M_A .
- Proportional to M_B .
- G = Newton's constant
 $= 6.67430(15) \times 10^{-11}$
 $m^3 / Kg \cdot s^2$

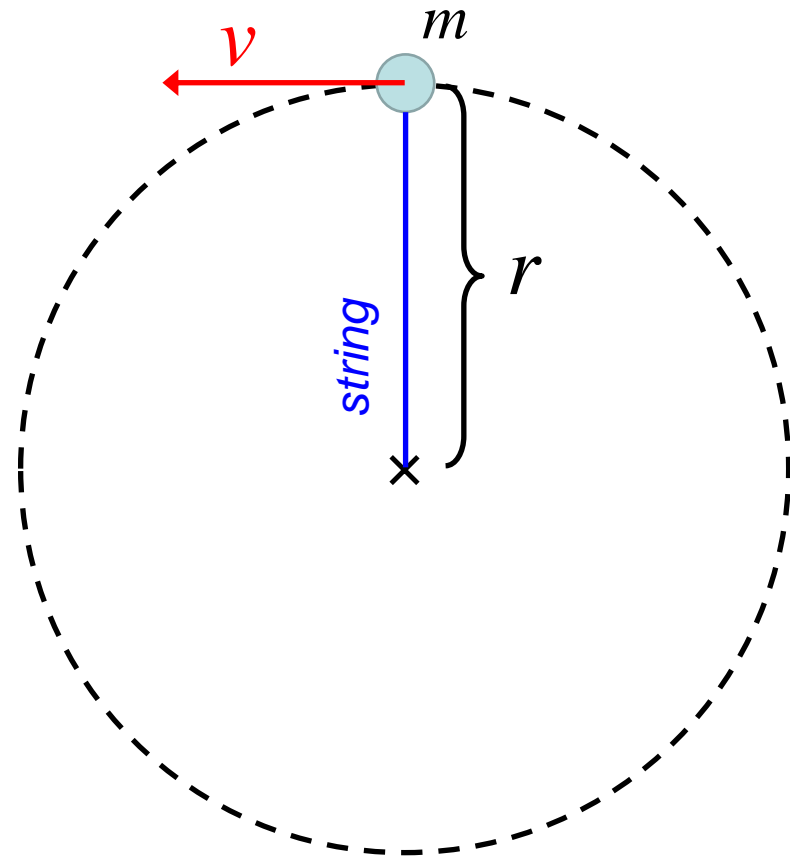
Why do all objects (on Earth)
fall
at the same rate?

Circular Motion

Recall

acceleration = **change** in velocity over time
speed & direction

Rotation is a type of acceleration
where the velocity **direction**
changes, but speed is constant.



“ball on a string”

Circular Motion

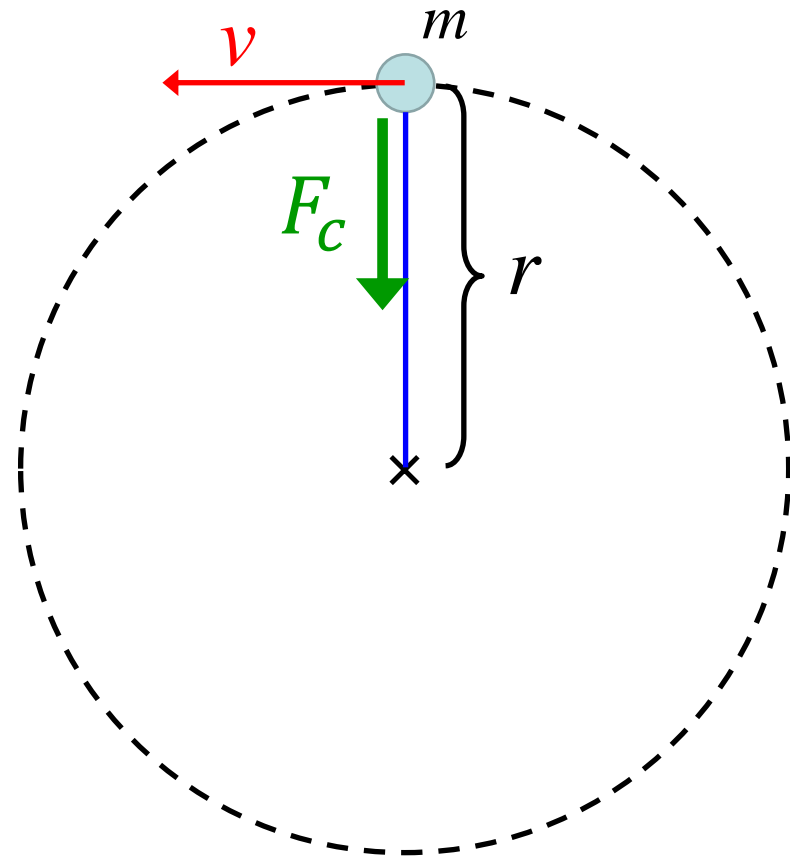
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Acceleration: $a_c = \frac{v^2}{r}$

Centripetal Force: $F_c = \frac{mv^2}{r}$

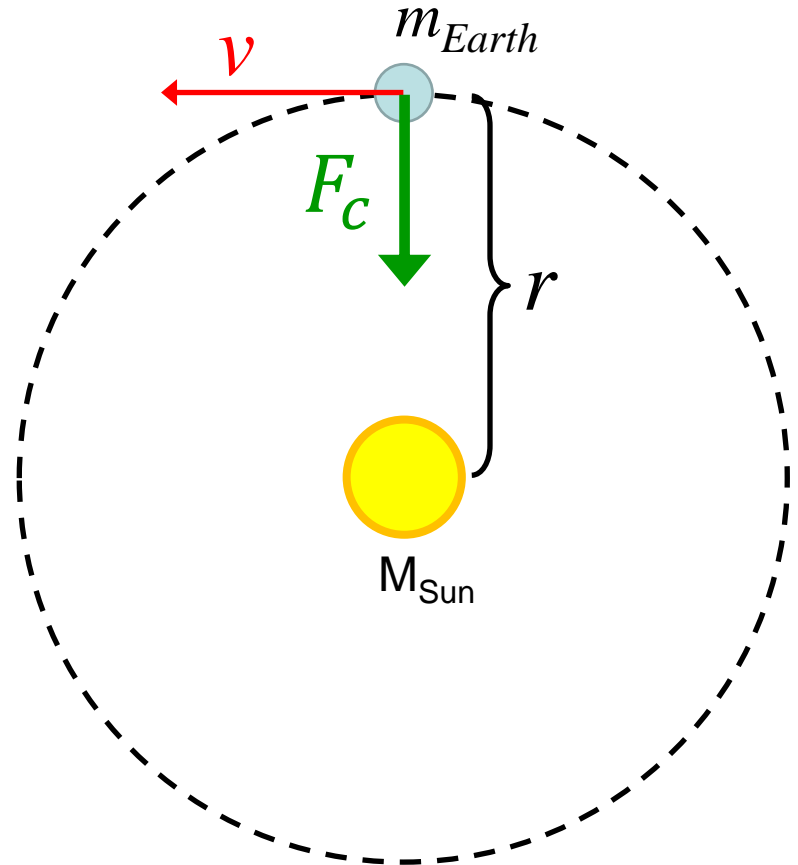


“ball on a string”

Circular Motion Example: Earth's orbit of Sun

Centripetal force needed to keep
Earth on a circular orbit:

$$F_c = \frac{m_{Earth} v_{Earth}^2}{r}$$



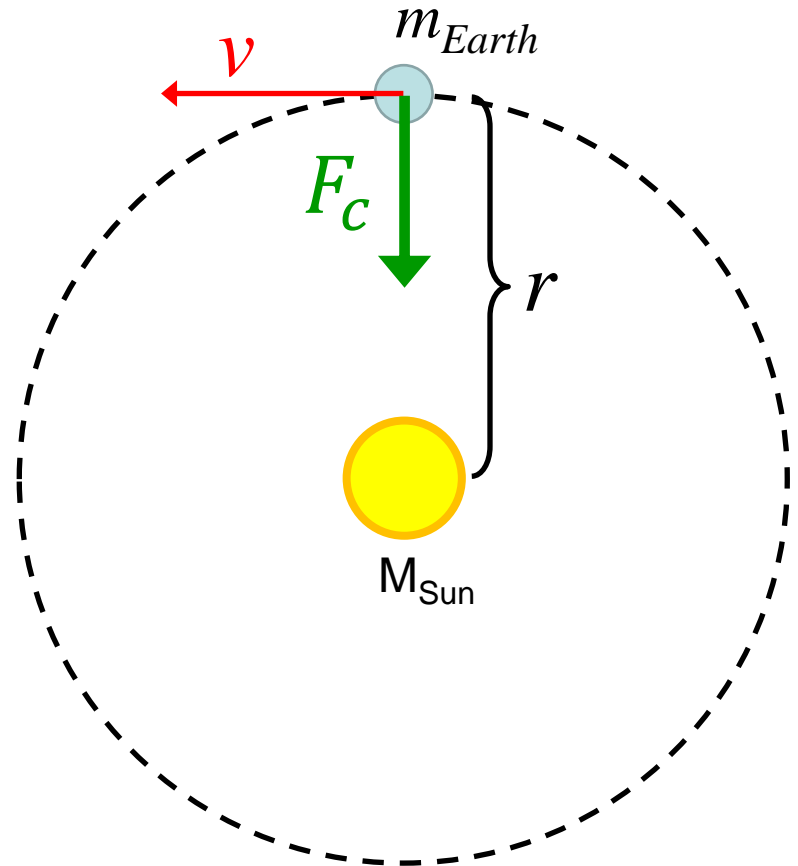
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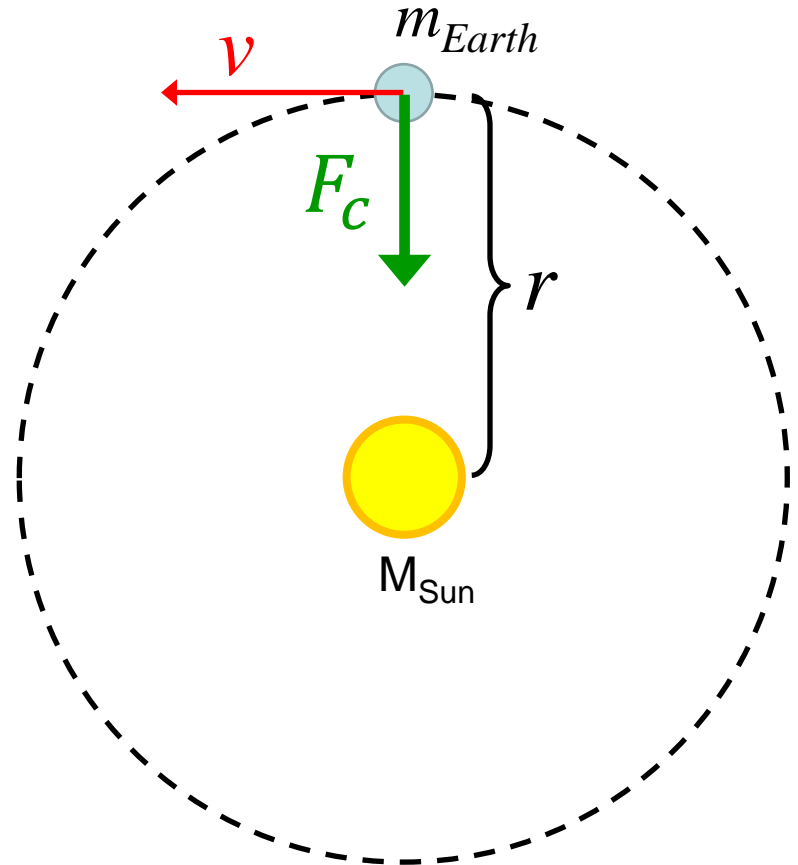
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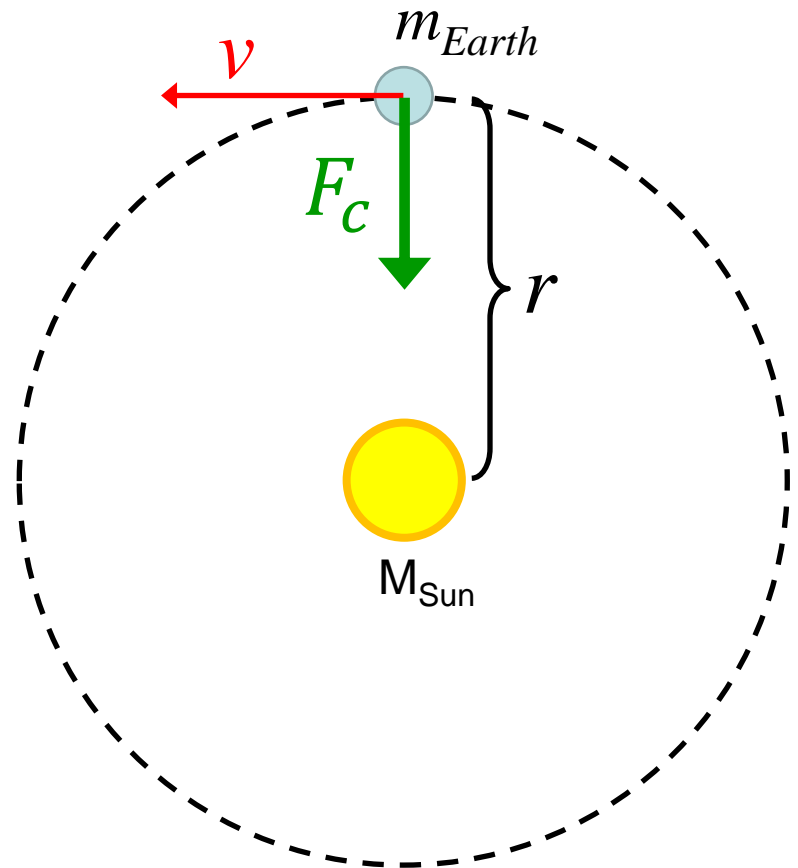
$$F_c = F_{gravity, S \rightarrow E}$$

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Circular Motion Example: Earth's orbit of Sun

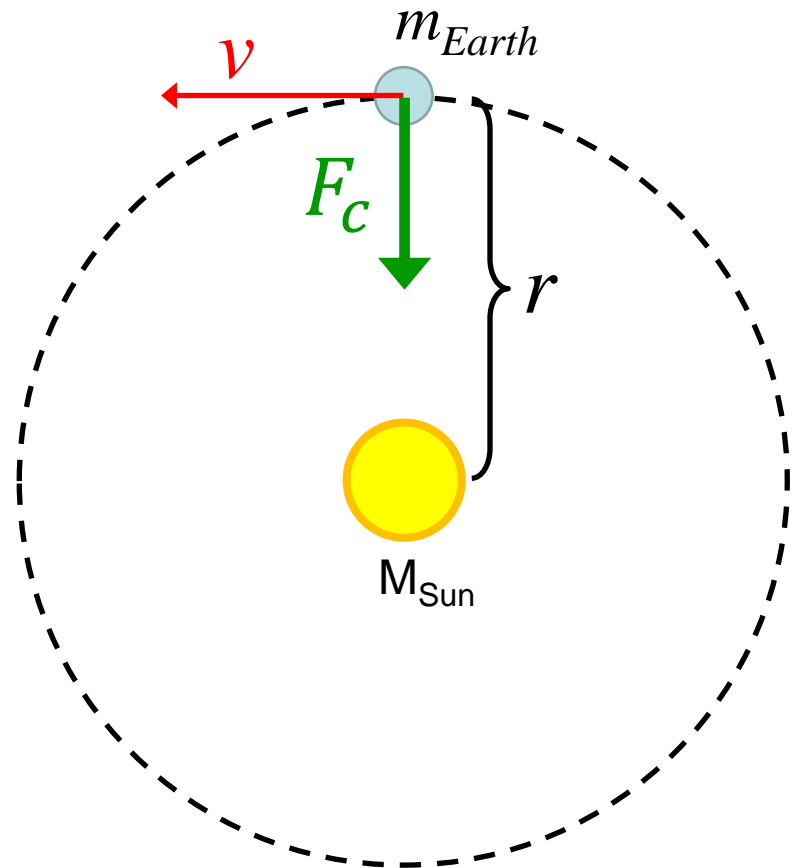
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Circular Motion Example: Earth's orbit of Sun

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$$\Leftrightarrow v_{Earth}^2 = G \frac{M_{Sun}}{r}$$



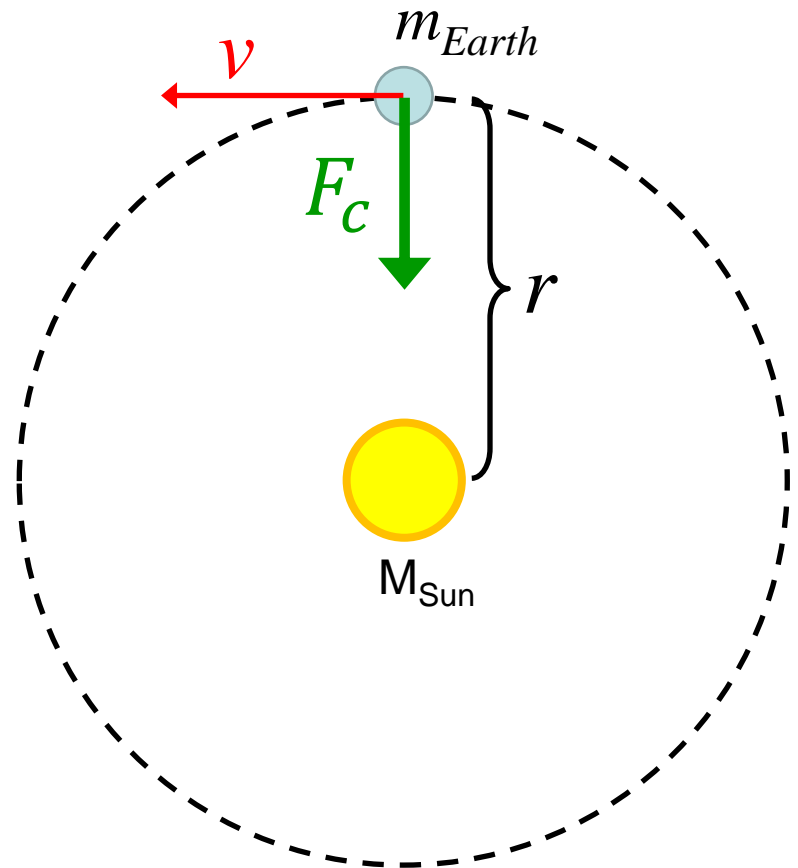
Circular Motion Example: Earth's orbit of Sun

$$\frac{\cancel{m_{Earth}} v_{Earth}^2}{r} = G \frac{\cancel{m_{Earth}} M_{Sun}}{r^2}$$

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Solve for M_{Sun} :

$$M_{Sun} = \frac{r v_{Earth}^2}{G}$$



Circular Motion Example: Earth's orbit of Sun

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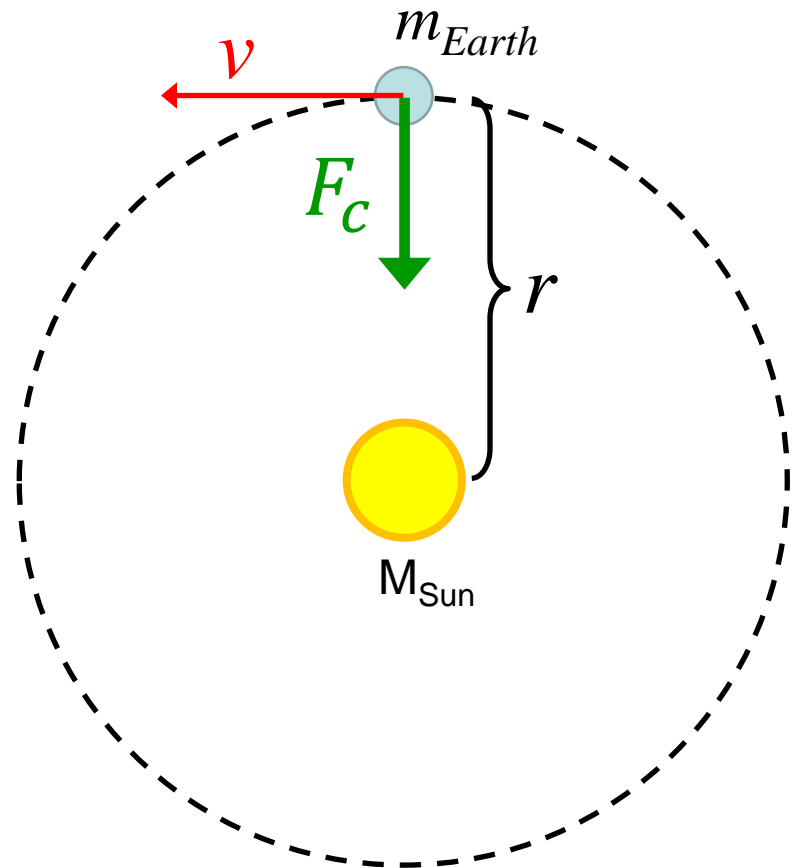
Solve for M_{Sun} :

$$M_{Sun} = \frac{r v_{Earth}^2}{G}$$

$$v_{Earth} = 29.78 \times 10^3 \text{ m/s}$$

$$r = 1 \text{ AU} = 149.6 \times 10^9 \text{ m}$$

$$G = 6.67430(15) \times 10^{-11} \text{ m}^3/\text{Kg} \cdot \text{s}^2$$



Circular Motion Example: Earth's orbit of Sun

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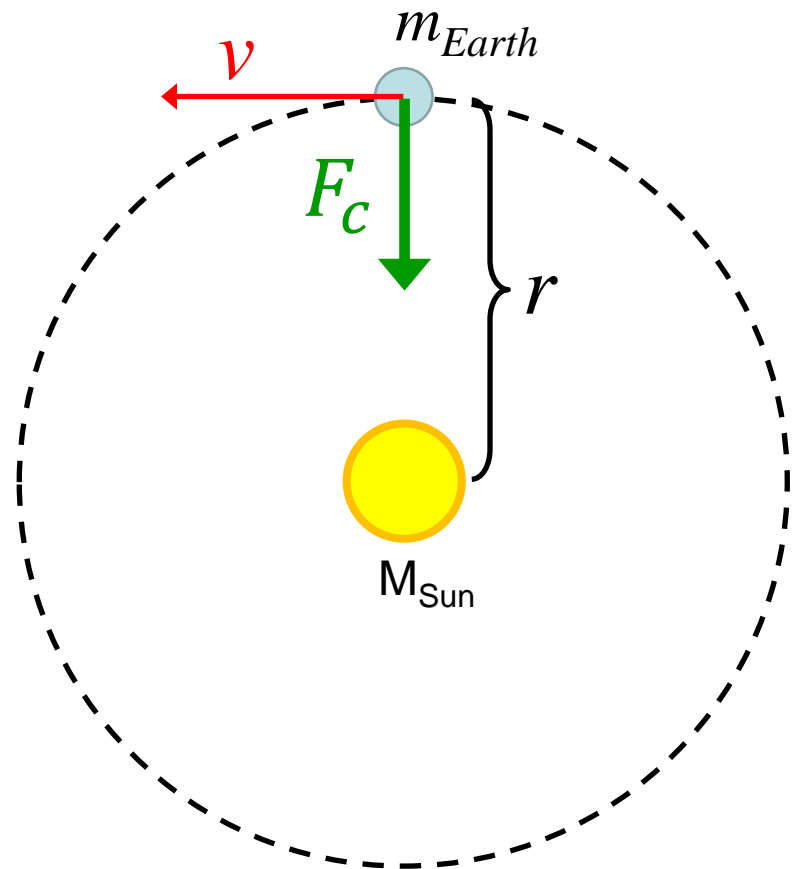
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$$M_{Sun} = 1.988 \times 10^{30} \text{ Kg}$$



You can get the mass of the Sun from
Earth's orbital parameters !!!

Newton's version of Kepler's 3rd Law

$$T^2 = \frac{4\pi^2}{G(M_1 + M_2)} a^3$$

This formula is in SI units

T = orbital period in seconds

$M_{1,2}$ = Mass of orbiting objects in Kg

a = semimajor axis in meters

$G = 6.6743 \times 10^{-11} \text{ m}^3/\text{Kg.s}^2$

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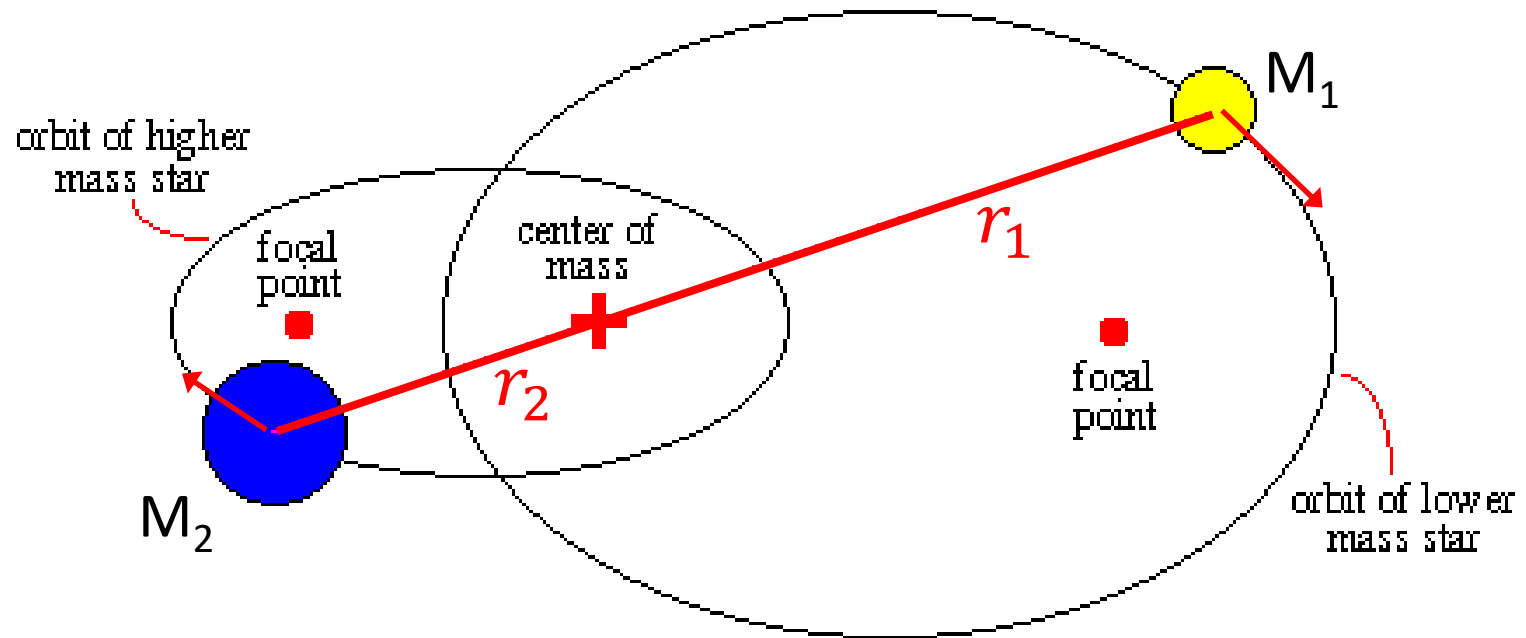
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WHAT IF: What happens to the orbits if M_1 and M_2 are comparable ?

Center of Mass

What happens when $M_1 \neq M_2$?

The **center of mass** of M_1 and M_2 serves as the orbiting ellipse focus.

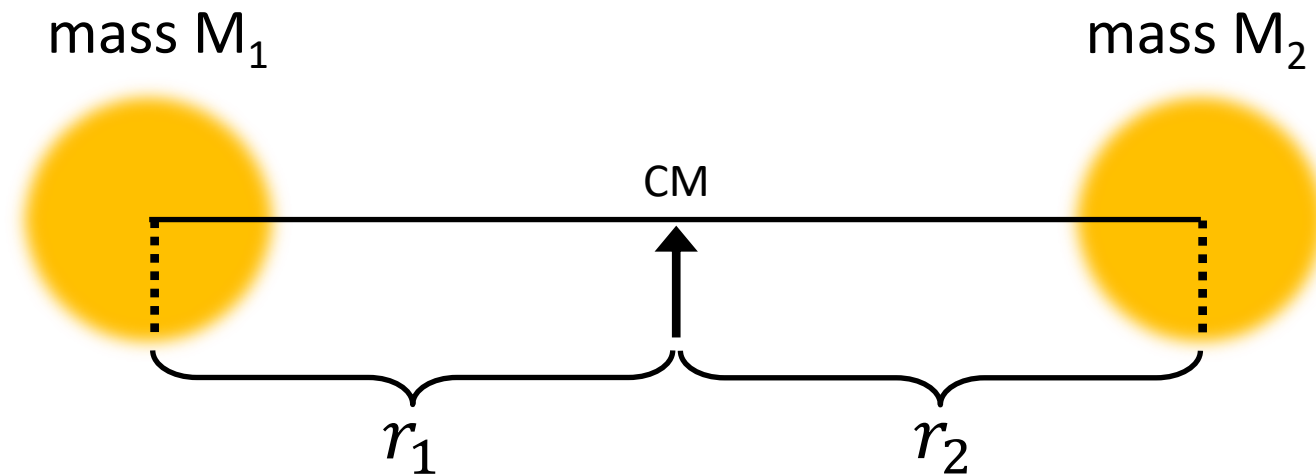


[adapted from <http://abyss.uoregon.edu>]

Semimajor axis "a":

The coordinate " $r = r_1 + r_2$ " is the distance between the two masses. It also describes an ellipse (not shown), whose semimajor axis "a" is used in Newton's version of Kepler's 3rd law.

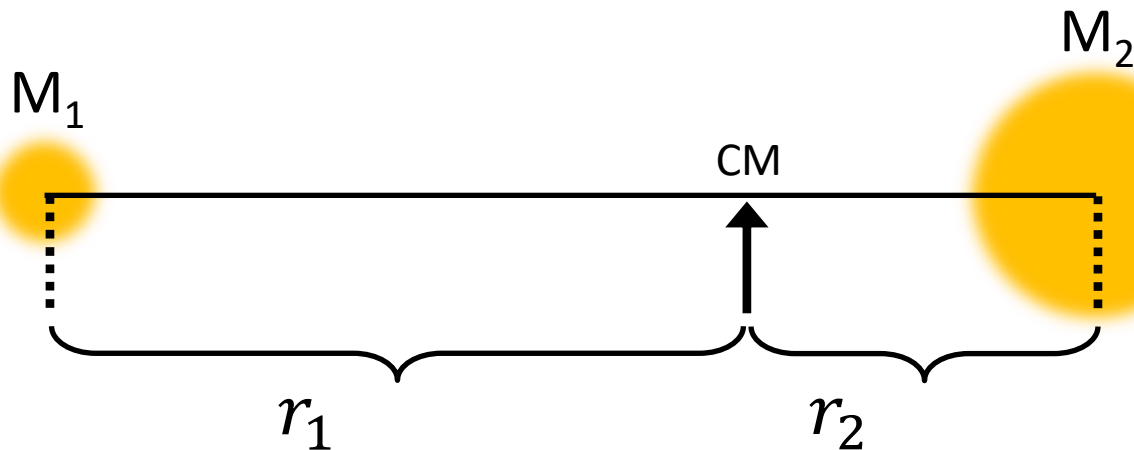
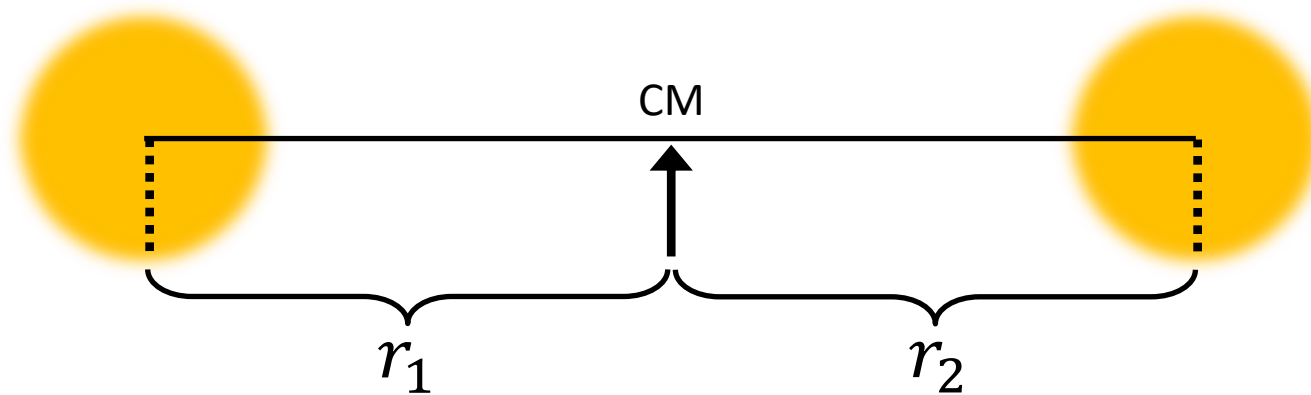
Center of Mass



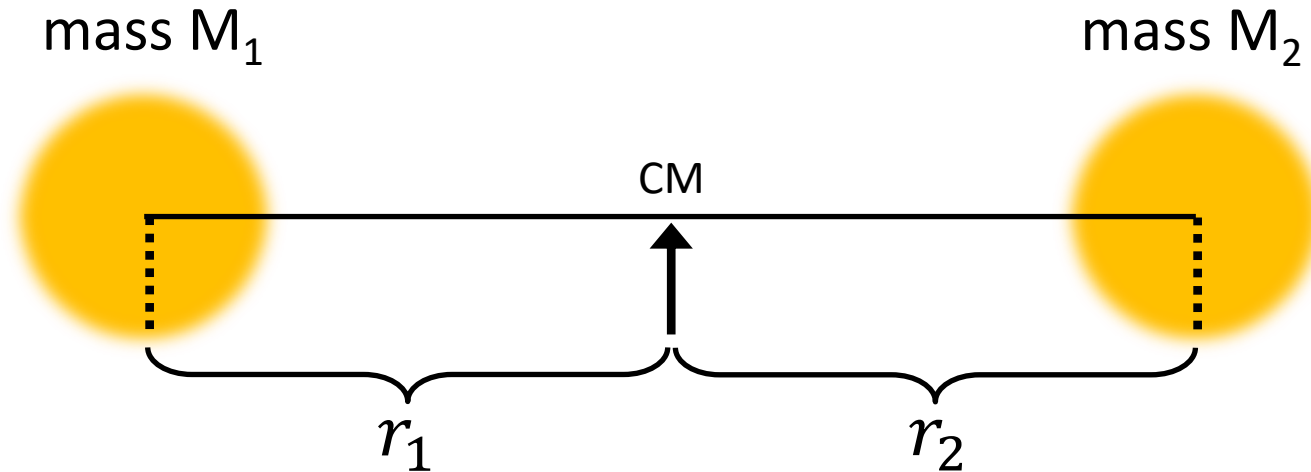
Center of Mass

mass M_1

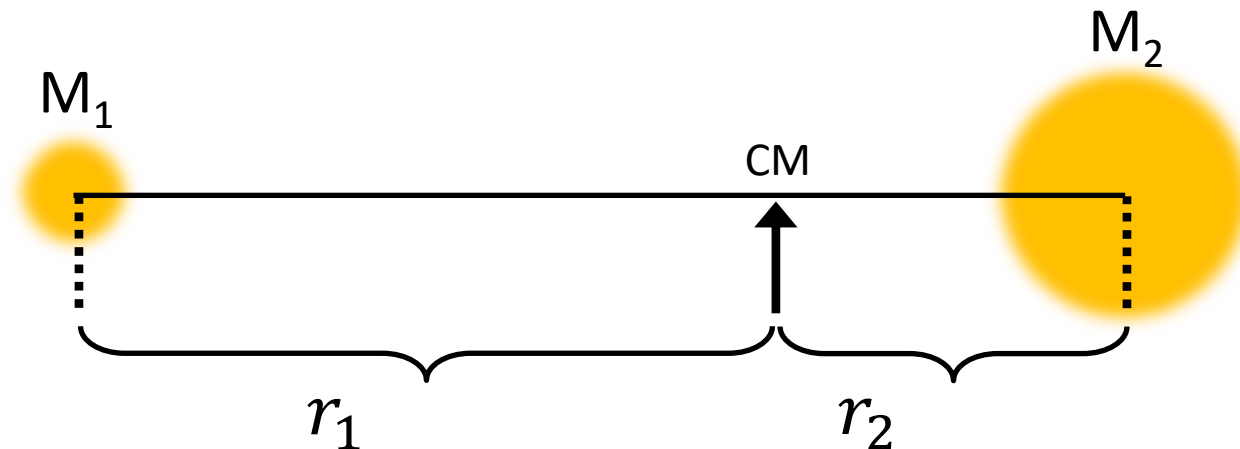
mass M_2



Center of Mass



Center of mass is located such that $M_1 r_1 = M_2 r_2$
(or "barycenter")



Some Barycenters

$M_2 - M_1:$ $r_2 = a \frac{M_1}{M_1 + M_2} = \text{distance from CM to } M_2$

Sun-Earth: $r_2 = 448 \text{ km} = 3.0 \times 10^{-6} \text{ AU}$

Earth-Moon: $r_2 = 4,670 \text{ km}$ with $a = 384,000 \text{ km}$
 $= 73\% \text{ of Earth's radius}$

Some Barycenters

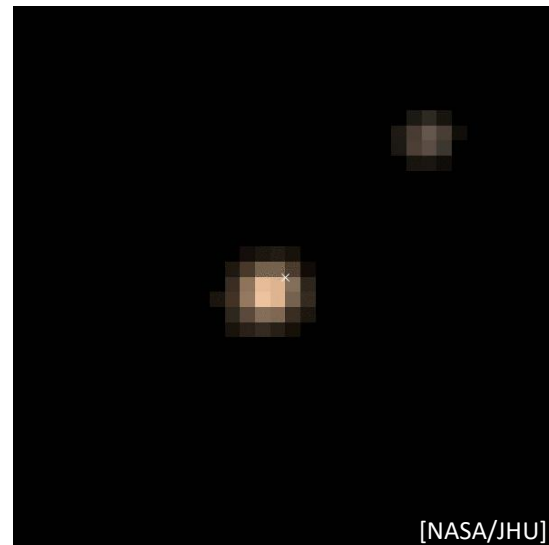
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Pluto – Charon:

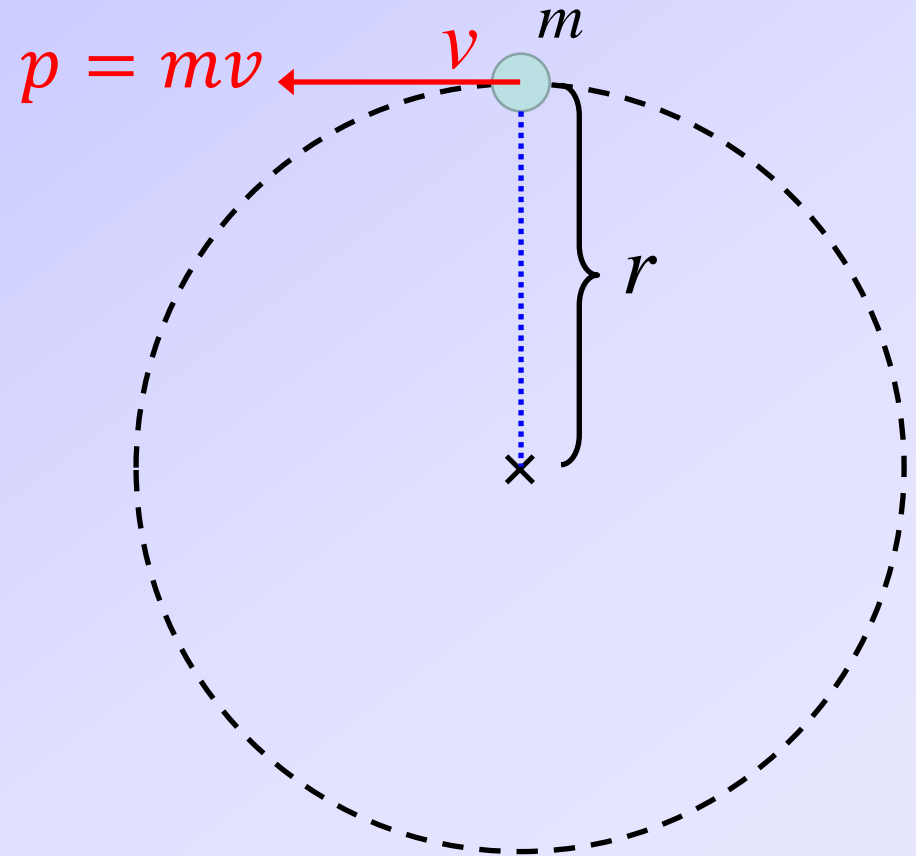
orbital period $T = 6.4$ days



PollEv Quiz: PollEv.com/sethaubin

Conservation of Angular Momentum (1)

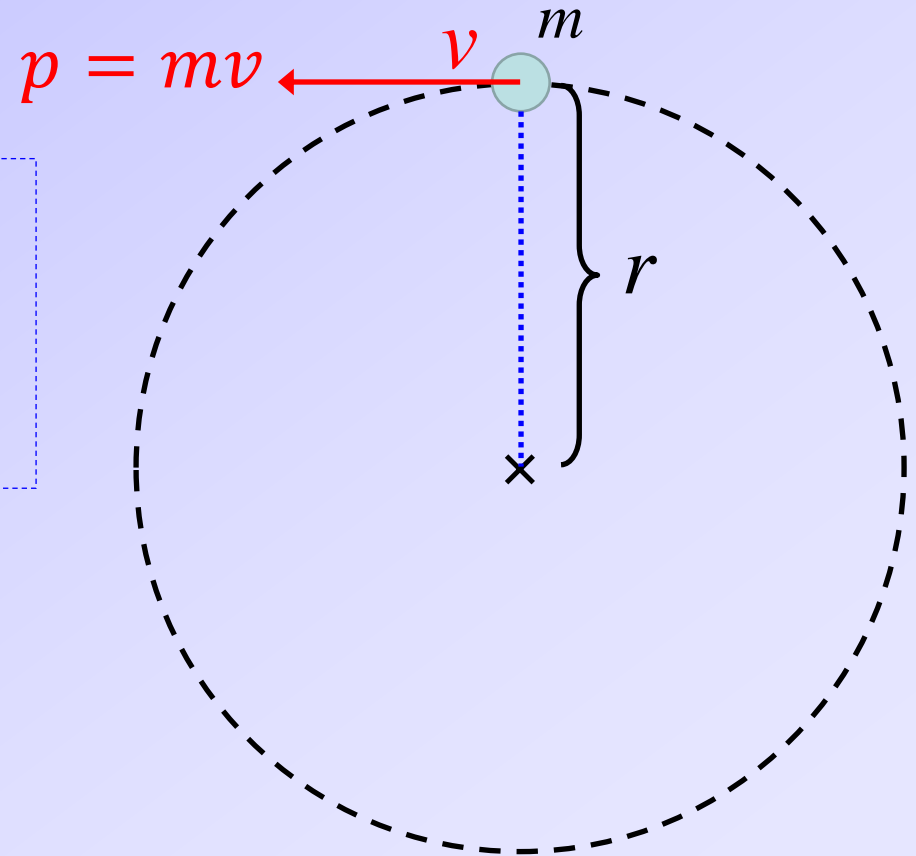
angular momentum = L = momentum \times radius
 $= p \times r \quad \dots = mvr$ for circular motion



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total angular momentum
=
sum of the angular momenta of
all the sub-parts of a system



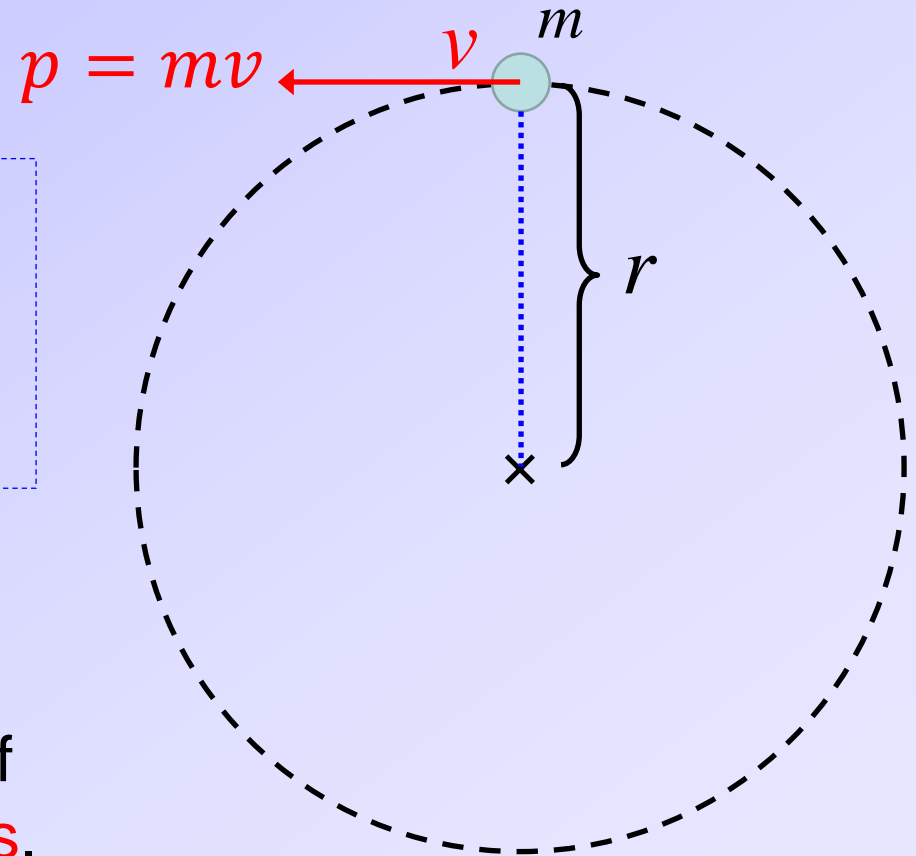
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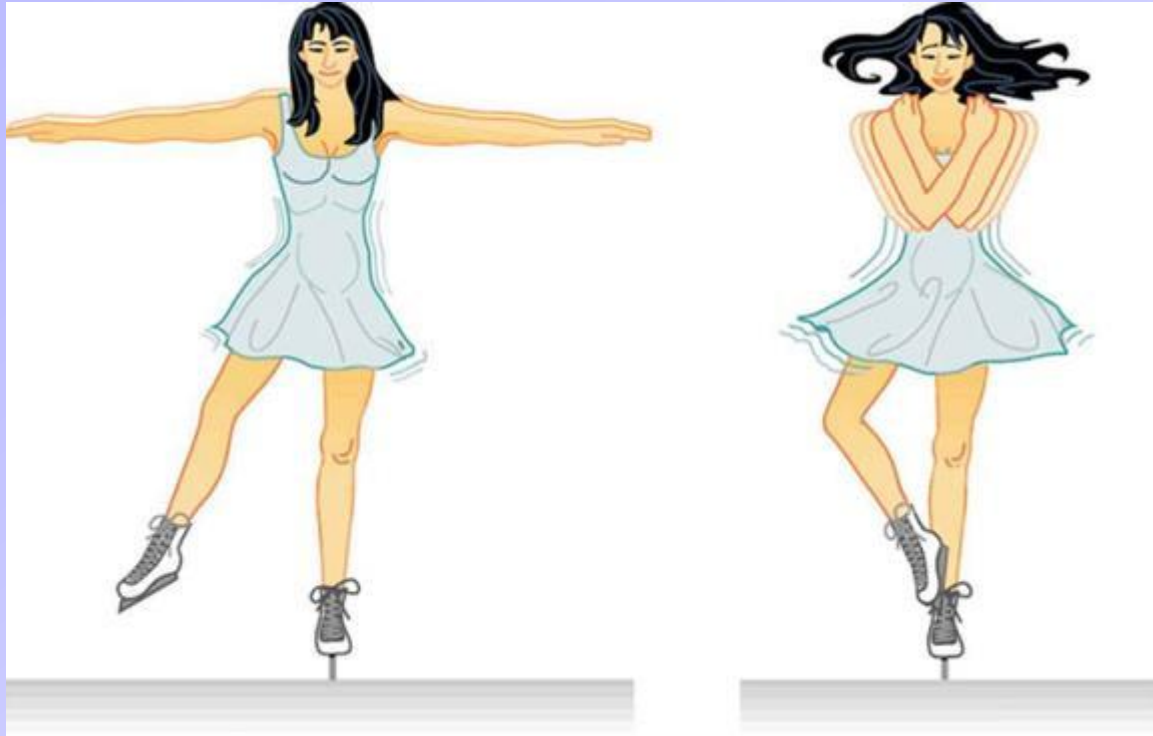
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Conservation Law

The **total angular momentum** of
a **closed system** **never changes**.



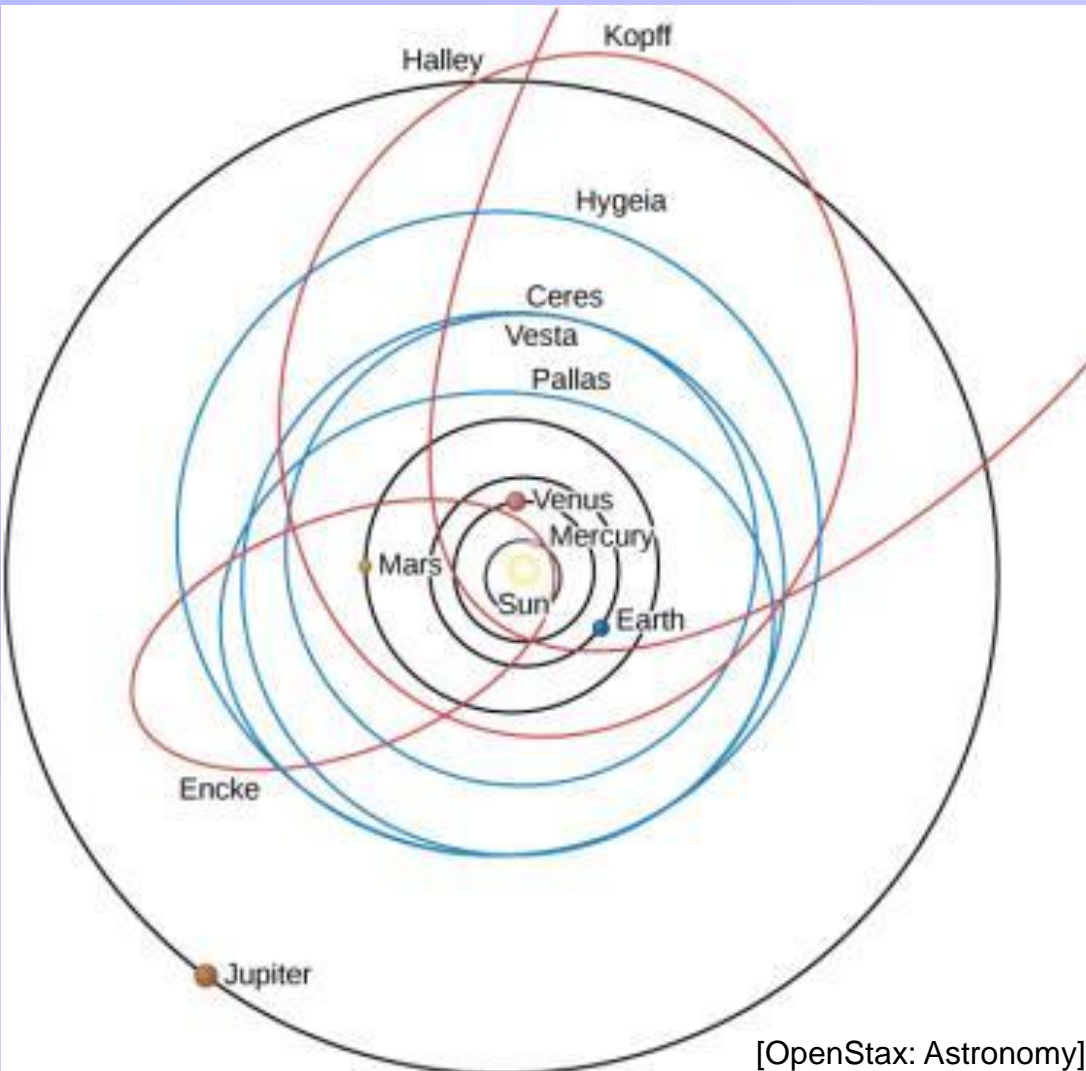
Conservation of Angular Momentum (2)



[OpenStax: Astronomy]

- When a spinning figure skater **brings in her arms**, their distance from her spin center is **smaller**, so her **speed increases**.
- When her **arms are out**, their distance from the spin center is **greater**, so she **slows down**.

Conservation of Angular Momentum



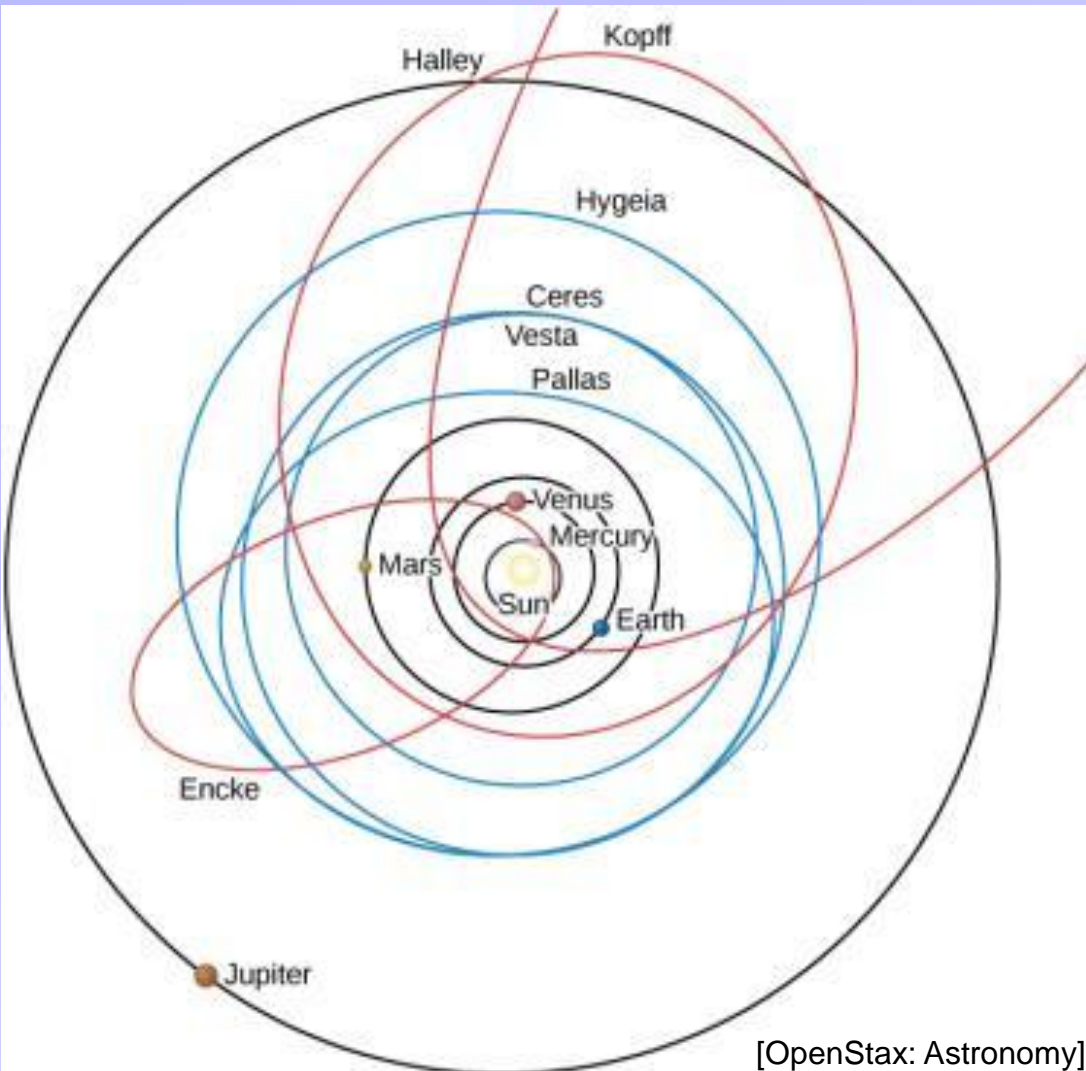
The multiple planets, asteroids, and comets all interact and modify each others orbits.

→ **Individual angular momenta change.**

→ **Total angular momentum of Solar System is constant.**

Planets (black), asteroids (blue), comets (red)

Conservation of Angular Momentum



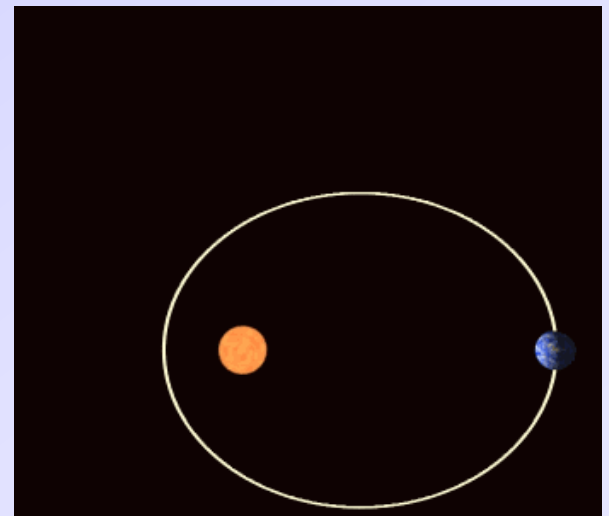
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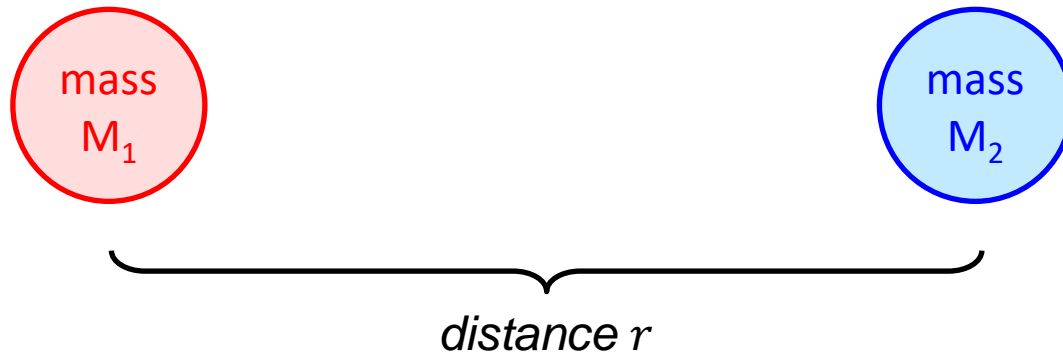
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Example: Apsidal Precession



Gravitational Potential Energy



$$\text{Stored gravitational energy} = E_{\text{potential}} = -G \frac{M_1 M_2}{r}$$

$$\text{Total Energy} = E_{\text{total}} = E_{\text{potential}} + E_{\text{kinetic}}$$

For 2 orbiting bodies (e.g. Sun + Earth): $E_{\text{total}} < 0$

For 2 unbound bodies (Earth + Mars rocket): $E_{\text{total}} > 0$

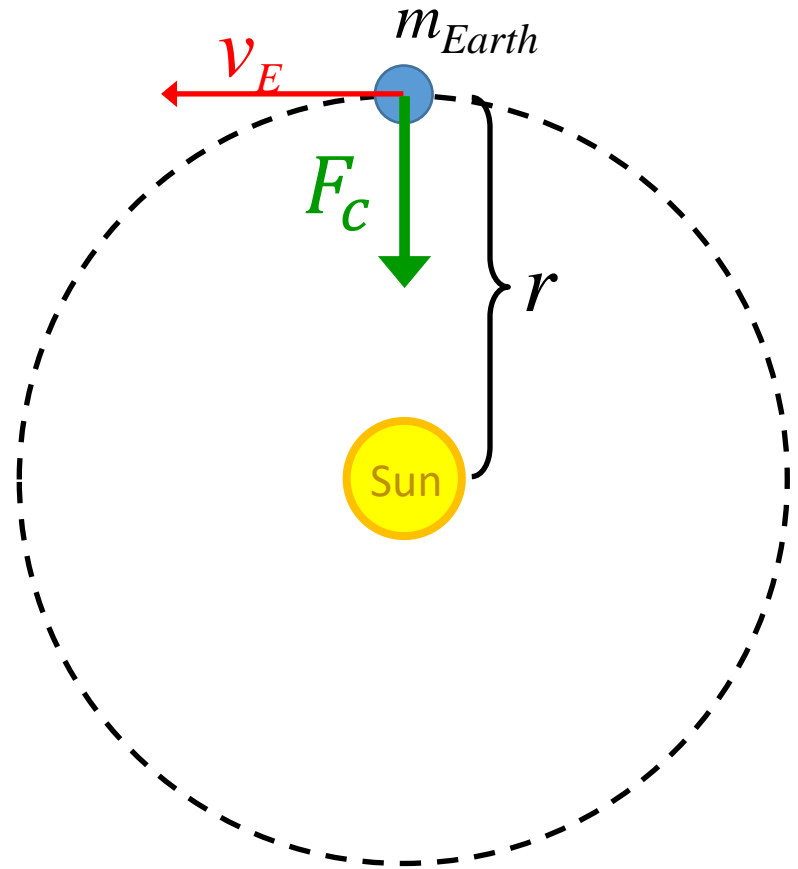
Bound Orbital Energy

Centripetal force needed to keep Earth on a circular orbit:

$$F_c = \frac{m_{Earth} v_{Earth}^2}{r}$$

Force of gravity on Earth from Sun:

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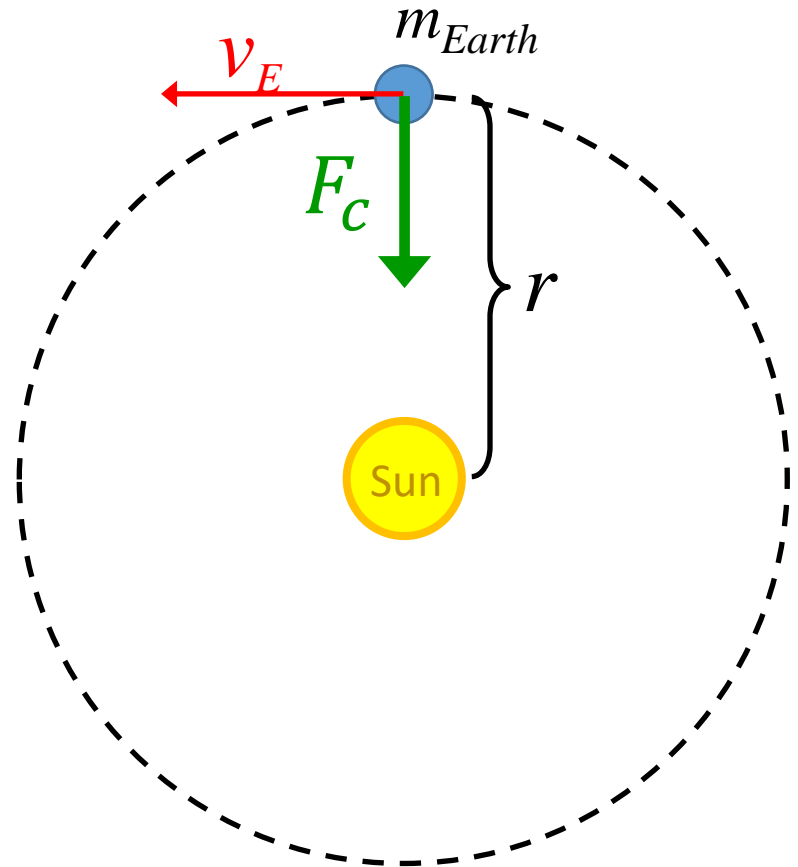
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The **centripetal force** that pulls on Earth to make it orbit the Sun **is gravity**:

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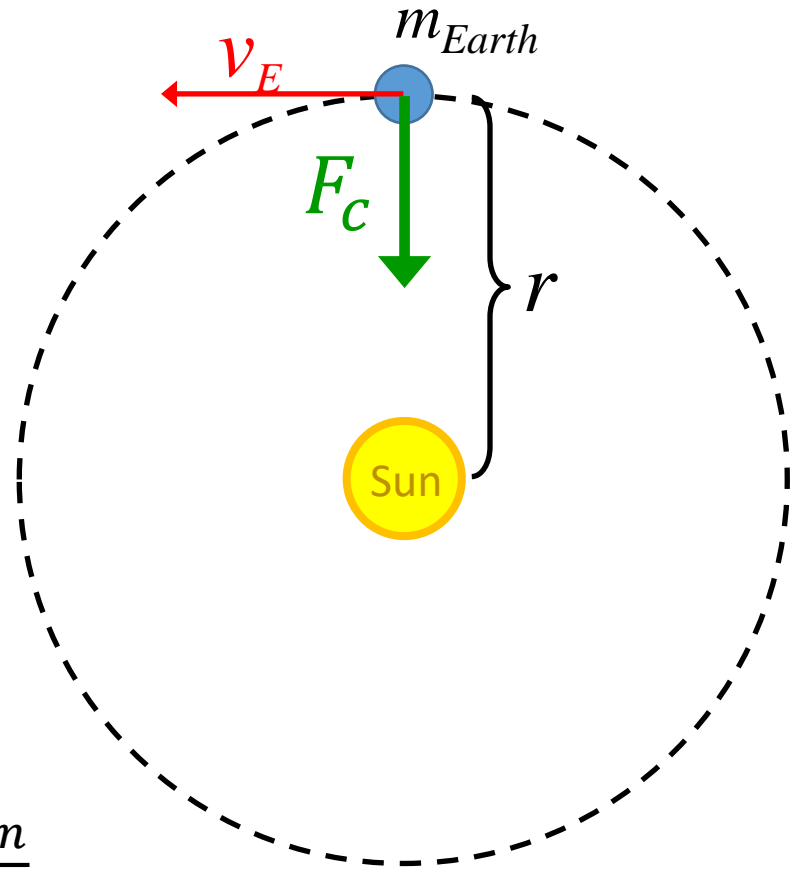
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$$\Leftrightarrow \frac{m_{\text{Earth}} v_{\text{Earth}}^2}{r} = G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r^2}$$

$$\Leftrightarrow \frac{1}{2} m_{\text{Earth}} v_{\text{Earth}}^2 = \frac{1}{2} G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r}$$



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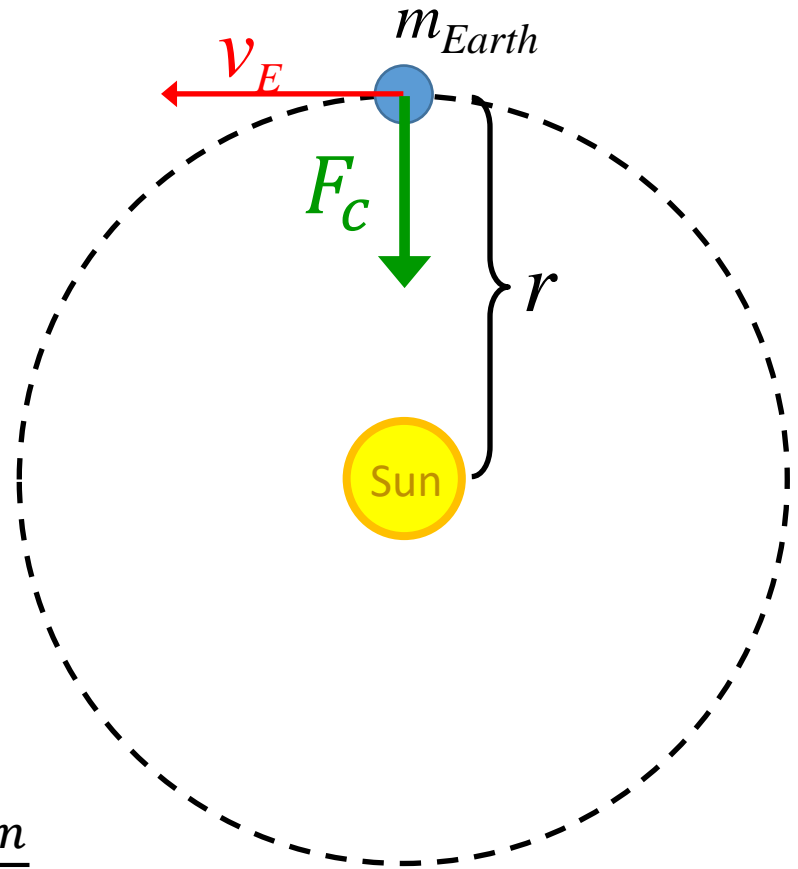
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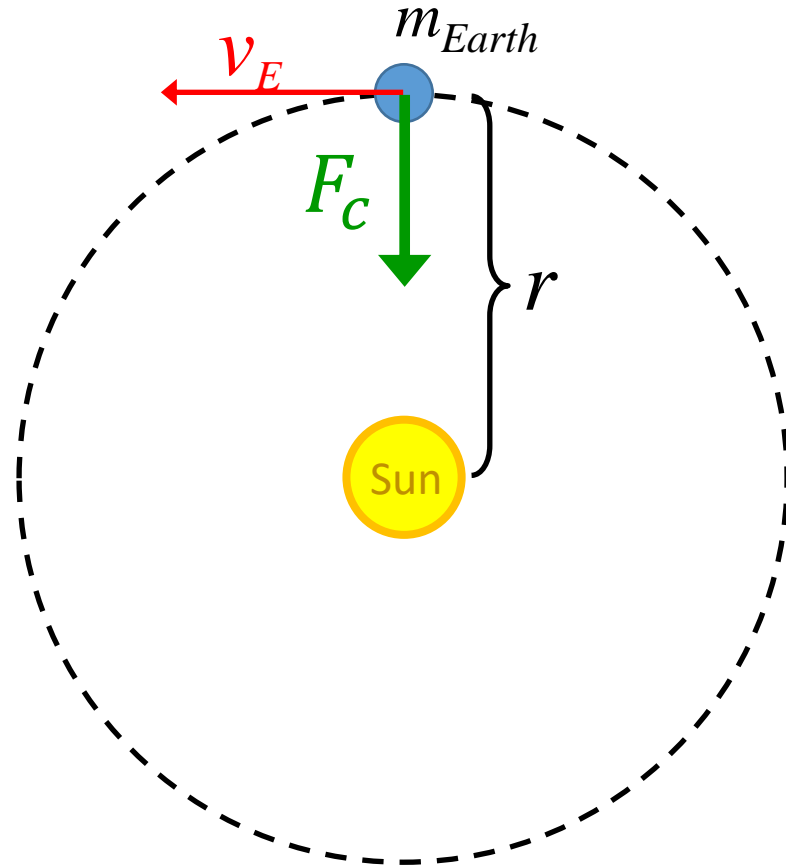
$$\Leftrightarrow \frac{1}{2} m_{\text{Earth}} v_{\text{Earth}}^2 = \frac{1}{2} G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r}$$

$$\Leftrightarrow E_{\text{kinetic}} = \frac{1}{2} m_{\text{Earth}} v_{\text{Earth}}^2 = \frac{1}{2} G \frac{m_{\text{Earth}} M_{\text{Sun}}}{r}$$



Bound Orbital Energy

$$E_{kinetic} = \frac{1}{2} G \frac{m_{Earth} M_{Sun}}{r}$$

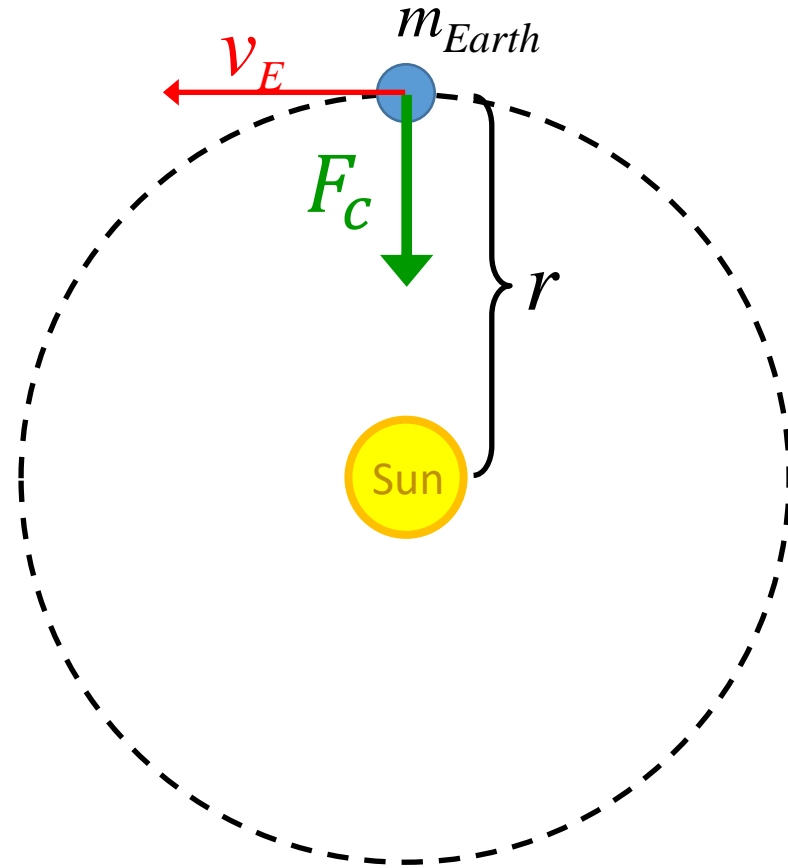


Bound Orbital Energy

$$E_{kinetic} = \frac{1}{2} G \frac{m_{Earth} M_{Sun}}{r}$$

Total Energy:

$$E_{Total} = E_{kinetic} + E_{potential}$$

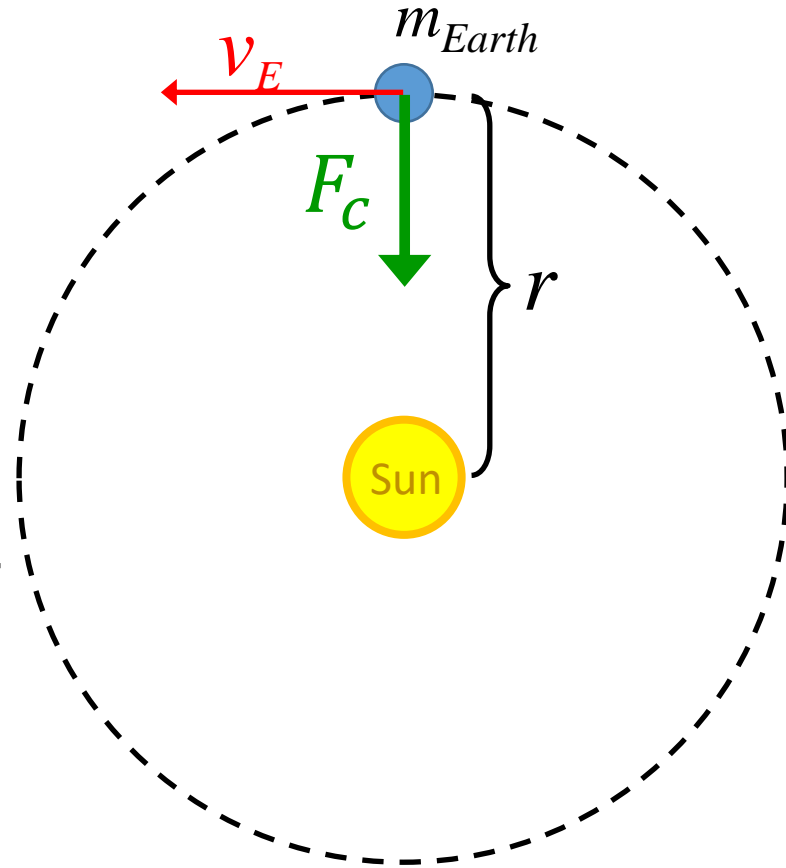


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$$\begin{aligned} E_{Total} &= E_{kinetic} + E_{potential} \\ &= \frac{1}{2} G \frac{m_{Earth} M_{Sun}}{r} - G \frac{m_{Earth} M_{Sun}}{r} \end{aligned}$$



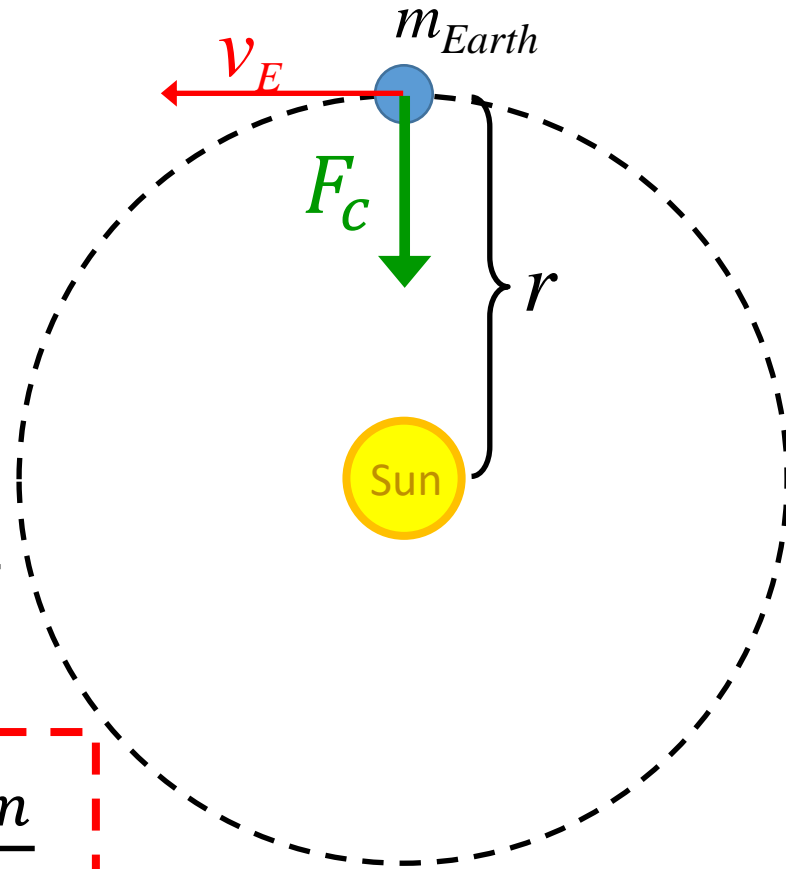
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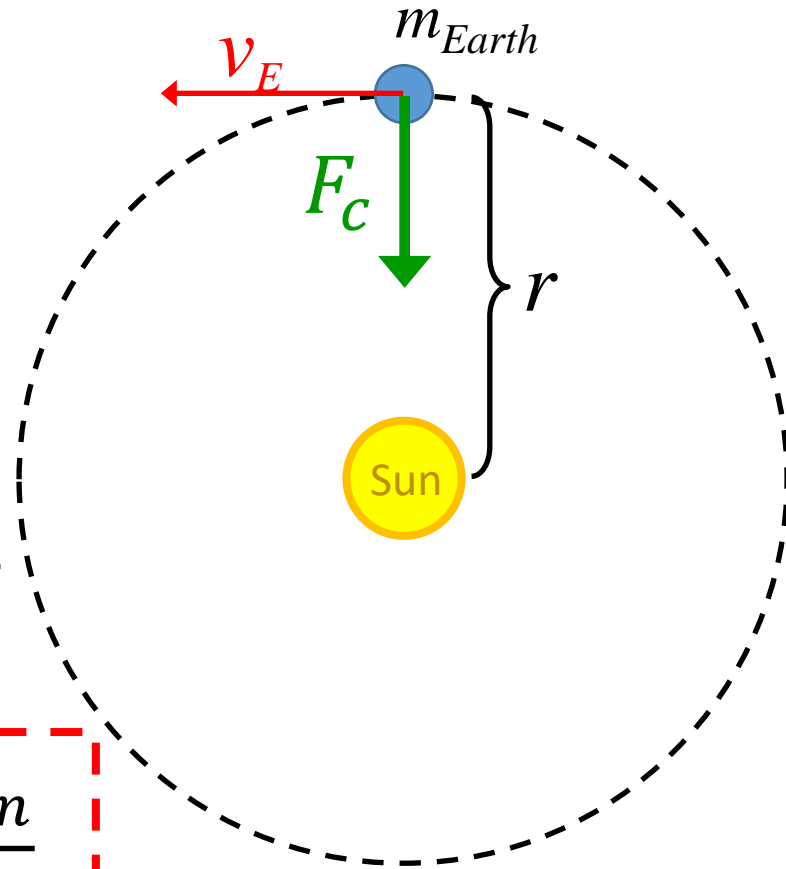
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The bound orbital energy is negative: $E_{Total} < 0$

Example: When a rocket wants to orbit another planet it has to slow (lower its energy) in order to go into orbit.